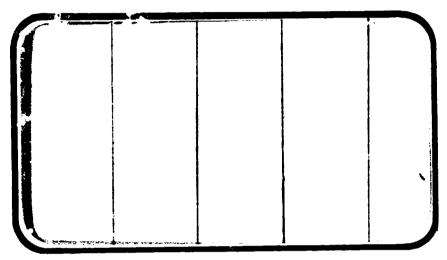


NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



(NASA-CR-147608) AN INVESTIGATION TO DETERMINE THE PRESSURE DISTRIBUTION OR THE 0.0137 SCALB SOLID BOCKET BOCSTER FOREBODY (MSPC MODEL 467) AT ABGLES OF ATTACK AT OR NEAR 90 DEG AND HIGH FRYNOLDS NUMBERS IN THE G3/18 45766

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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT



JOHNSON SPACE CENTER HOUSTON, TEXAS

DATA MANagement services SPACE DIVISION

DMS-DR-2207
NASA-CR-147,608
AN INVESTIGATION TO DETERMINE

THE PRESSURE DISTRIBUTION ON THE 0.0137 SCALE

SOLID ROCKET BOOSTER FOREBODY

(MSFC MODEL 467) AT ANGLES OF ATTACK AT
OR NEAR 90° AND HIGH REYNOLDS NUMBERS IN THE
MSFC HIGH REYNOLDS NUMBER WIND TUNNEL (SA29F)

by

Paul E. Ramsey, NASA/MSFC

Prepared under NASA Contract Number NAS9-13247

by

Data Management Services Chrysler Corporation Space Division New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center National Aeronautics and Space Administration Jouston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: MSFC HRWT 033

NASA Series No.: SA29F Model Number: 467

Test Dates: August 8 - September 18, 1974

June 25 - July 14, 1975

FACILITY COORDINATOR:

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Chrysler Corporation Space Division assumes no responsibility for the data presented other than display characteristics.

AN INVESTIGATION TO DETERMINE THE PRESSURE

DISTRIBUTION ON THE 0.0137 SCALE SOLID ROCKET

BOOSTER FOREBODY (MSFC MODEL 467) AT ANGLES

OF ATTACK AT OR NEAR 90° AND HIGH REYNOLDS NUMBERS

IN THE MSFC HIGH REYNOLDS NUMBER WIND TUNNEL (SA29F)

by

Paul E. Ramsey, NASA/MSFC

ABSTRACT

An aerodynamic investigation was conducted in the MSFC High Reynolds Number Wind Tunnel to determine the pressure distribution over the foresection of the current 146 inch diameter Shuttle SRB. The test model consisted of a 0.0137 scale version of the SRB nose cone and a forward portion of the cylindrical body which was approximately 2.7 calibers in length. The external configuration was defined by MSFC drawing number 10A00313 Revision A.

Data were obtained for Mach numbers of 0.4, 0.5, 0.6, C.8, and 2.0; Reynolds numbers of from $2.3(10)^6$ to $20.1(10)^6$ based on body diameter; angle of attack of 72° , 90° , and 108° , circumferential body locations from 0° to 180° ; and longitudinal station ratioed to body diameter (X/D) of from 0.340 to 2.160. Yaw was zero and no protuberances were fitted.

The pressure distributions are plotted as a function of longitudinal station ratioed to body diameter and circumferential location for

ABSTRACT (Concluded)

each angle of attack and Mach number. A Reynolds number variation study was made for Mach numbers of 0.4 and 0.6 at an angle of attack of 270° and roll angle of 180° .

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	CONDITIONS VARYING	THETA MACH ALPHA	X/D MACH ALPHA	RN X/D ALPHA MACH
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PLOTTED COEFFICIENTS SCHEDULE:

(A) CP versus X/D

(B) CP versus THETA

NOMENCLATURE

General

SYMBOL	PLOT SYMBOL	DEFINITION	UNITS
C _p	СР	pressure coefficient; $(p_1 - p_{\infty})/q_{\infty}$	
D		reference diameter (body diameter)	
₽B	LBODY	length of the ET	in.
M		Mach number	
P1		local pressure	psi
Pt		total pressure	psi
P _∞		freestream pressure	psi
q_{∞}	Q(PSI)	freestream dynamic pressure unit	psi
R _N	RN	Reynolds number based on model diameter	
RN/L	RN/L	Reynolds number per unit length	
SRB		Solid Rocket Booster	
S _{Ref}	SREF	base area or reference area, ft ²	
Tt		freestream total temperature (°F)	deg.
X		distance from nose of SRB along X axis	in.
X_{T},Y_{T},Z_{T}		SRB stations	in.
X/D	X/D	longitudinal location of pressure measurement, expressed as a fraction of the SRB length, measured from the SRB nose	
Po		plenum chamber pressure	psia

NOMENCLATURE (Concluded)

SYMBOL	PLOT SYMBOL	DEFINITION	UNITS
GREEK SY	MBOLS		
α	ALPHA	angle of attack	deg.
ф	PHI	angle of roll	deg.
θ	THETA	circumferential location	deg.
SUBSCRIP	<u>rs</u>		
ref		reference conditions	
∞		freestream conditions	
1		local	
t		total conditions	

INTRODUCTION

During the descent of the SRB, an atmospheric pressure switch or baroswitch will be used to initiate the Shuttle SRB parachute deployment system. This device will be located in a forward compartment which will be vented to the exterior environment. The internal pressure of this compartment is a function of the external conditions such as SRB velocity, attitude, and altitude which are predicted from aerodynamic force inputs into a dynamics analysis program. Based on the internal pressure, which corresponds to the desired parachute deployment altitude for a given entry trajectory, the baroswitch can be preset for recovery system initiation at the proper time. Initiation must occur within narrow altitude limits to assure the occurrence of the dynamic pressure needed for proper functioning of the parachute system. To obtain an accurate prediction of the forward compartment pressure needed to set the baroswitch, the external pressure distribution must be known for the full scale flight conditions.

Since the SRB will enter at an angle of attack near 90° where the flow around the body can be very sensitive to Reynolds number, a test Reynolds number as near to full scale as possible is necessary if adequate data are to be obtained.

Based on these observations, a test was conducted in the MSFC High Reynolds Number Wind Tunnel to determine the SRB forebody pressure distribution at 72° to 108° angle of attack. Pressures were measured over the nose cone and approximately one caliber of the cylindrical body. The

INTRODUCTION (Concluded)

Mach range consisted of 0.4, 0.5, 0.6, 0.8, and 2.0 at Reynolds numbers based on the SRB diameter. No protuberances were fitted for this investigation.

TEST HARDWARE

The model configuration consisted of a spherically blunted cone/cylinder which simulated the forward portion of the current 146-inch-diameter SRB. The model was mounted at the aft ϵ 'with a 90° "dog leg" sting which was then supported in an available tunnel sting. Major dimensions of the model and sting are presented in Figure 2.

Because only 18 pressure tubes of 0.125-inch 0.D. could be accommodated in the model, 4 interchangeable nose sections were provided, each containing 18 pressure orifices. The pressure tubing passed out of the nose through an 0-ring sealed quick-disconnect system which allowed easy interchangeability of the nose sections. From this disconnect the tubing passed through cavities in the sting to the tunnel exterior.

No protuberances were included on the test models. The model diameter was 2 inches, which resulted in a scale of 0.0137. The MSFC model number is 467 and was fabricated using drawings 80M42699, 80M42700, 80M42714 - 80M42⁻¹⁷, and 80M51377.

INSTRUMENTATION

The orifice location and arrangement are shown in Figure 3. The pressure lines were tubed to 18 Statham 1000 psia pressure transducers, which were monitored real time during each run. Combining data from each noise resulted in a total set of data for 72 orifice locations distributed over the test SRB configuration. In effect, four blows or runs were required to obtain a complete set of data for each test condition; i.e., Mach number, α , P_0 , and Reynolds number.

Two check orifices at thetas of 202½° and 337½° at longitudinal station 7 were included on nose 4 to give an indication of model misalignment or flow asymmetries. Because of the limited interior volume available near the front of the model, pressure taps were located on both sides of the nose for longitudinal station 1 on nose 4.

TEST FACILITY DESCRIPTION

The MSFC High Reynolds Number Wind Tunnel (HRWT) is a Ludwieg tube tunnel capable of producing a Reynolds number range of 7×10^6 to 200×10^6 per foot over a Mach number range of 0.25 to 3.50. The test section is 32 inches in diameter by 64 inches in length.

The supply tube has a 52-inch inside diameter and is 386 feet long. It is terminated on one end with a hemispherical head and on the other end with a stilling chamber. The stilling chamber is 20 feet long and has a maximum diameter of 78 inches. The entrance cone has an 8-degree included angle.

The facility uses six interchangeable, axisymmetric, contoured nozzles. The diameters at the entrance and exit of each nozzle are 52 and 32 inches, respectively. The "sonic" nozzle is used to produce all subsonic and transonic speeds. The other five nozzles are designed for discrete Mach numbers 1.4, 1.7, 2.0, 2.75, and 3.5. Further details and performance characteristics of the HRWTare presented in Reference 1.

DATA REDUCTION

The data were reduced to pressure coefficients as follows.

$$C_p = \frac{P_1 - P_{\infty}}{q_{\infty}}$$

and tabulated in the format shown in Table IV. The data were also formatted for inclusion in the CCSD data management system. Tunnel test condition parameters such as run number, Mach number, α , q_{∞} , T, RN, and P_0 are included in the tabulated data of the Appendix.

REFERENCES

 Gwin, H. S., "The George C. Marshall Space Flight Center High Reynolds Number Wind Tunnel Technical Handbook," NASA TM X-64831, December 1973.

TABLE I. TEST: MSFC HRWT 033 DATE: 10/17/75 TEST CONDITIONS MACH REYNOLDS DYNAMIC STAGNATION STAGNATION N JMBER NUMBER **PRESSURE TEMPERATURE** PRESSURE (per unit length) (pounds/sq.Inch) (degrees Fahrenheit) (pounds/sq.inch) 2.3 x 106 37 0.4 7.3 67.8 13.4 51 4.0 128.1 191.4 46 19.9 6.0 49 25.4 238.8 7.5 2*8*. Z 51 263.1 8.4 8.9 30.2 49 286.8 32.7 38 10.Z 315.9 44.6 48 13.3 424. Z 61.2 48 589.8 18.1 61.3 44 18.8 590.1 0.5 49 18.1 75.2 506.7 0.6 13.0 48 64.3 2.8 4.0 18.7 41 92.2 27.3 46 6.0 136.6 34.4 7.5 175.4 51 8.9 47 204.Z 40.3 BALANCE UTILIZED: NONE - STATHAM PRESSURE TRANSDUCERS

		CAPACITY:	ACCURACY:	COEFFICIENT TOLERANCE:
	NF	1000 PSI	± 5 PSI	±.06@q=83Psi
	SF			
	AF			
	PM		·	
	RM			
	YM			
COMMENTS:		-		

TABLE I. (Concluded) TEST: MSFC HRWT 033 DATE: TESI CONDITIONS MACH REYNOLDS DYNAMIC STAGNATION STAGNATION NUMBER PRESSURE TEMPERATURE PRESSURE NUMBER (per unit length) (pounds/sq.inch) (degrees Forrenhell) (pounds/sq.inch) 9.7 × 106 44.8 51 228.9 0.6 298.8 47 12.8 58.9 13.3 41 303.6 60.1 74.6 378.9 16.1 49 18.1 82.6 46 412.4 20.1 90.4 36 451.4 346.1 98.0 0.8 18.1 42 71.7 61 9.0 200-3 2.0 ٠, BALANCE UTILIZED: COEFFICIENT CAPACITY: ACCURACY: **TOLERANCE:** NF SF AF PM RM YM COMMENTS:

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TABLE II.	NC WWO N	-	(4)	(M) (M)	(B)(B		(S) (S)	,	(1) (C)												49	مديليت		
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TABLE II. (Continued)

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DENTIFIER	CONFIGURATION	Ö		ϕ	1.3	2,5	3 4.0	6.0	2.5	8.9	9.7	10.2	377	13.3	79/	18.8	20.1
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					ľ	COEFF	COEFFICENTS	S						100	10VAR (1)	IDVAR (2)	70N (2)
C OR C									•								
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TABLE II (Continued)

DATAMAN RUN NUMBER CORRELATION WITH TUNNEL RUN NUMBER

New Run Number	Tunnel Number	α	M
1	24/0, 19/0, 13/0, 18/0	72	0.4
2	1/0, 6/1, 12/1, 7/0	90	
3	25/0, 30/1, 36/0, 31/1	108	4
4	57/0, 83/0	72	0.5
5	59/1, 79/0	80	
6	63/1, 77/2	90	
7	65/1, 73/0	100	
8	69/0, 71/3	108	\
9	23/2, 20/0, 14/0, 17/0	72	0.6
10	60/1, 80/0	80	
11	2/0, 5/0, 11/1, 8/0	90	
12	66/1, 74/0	100	
13	26/0, 29/1, 35/0, 32/0	108	\
14	22/2, 21/0, 15/0, 16/0	72	0.8
15	61/3, 81/0	80	
16	3/0, 4/0, 10/3, 9/0	90	
17	67/2, 75/0	100	
18	27/0, 28/0, 34/0, 33/0	108	Ť
19	58/0, 84/0	72	2.0
20	62/0, 82/0	80	2.0

TABLE II. (Concluded)

New Run Number	Tunnel Number	α	M
21	64/2, 78/0	90	2.0
22	68/0, 76/1	100	2.0
23	70/0, 72/0	108	2.0

TABLE III. MODEL DIMENSIONAL DATA

MODEL COMPONENT: BODY - SRB FO	RWARD SECTION		
GENERAL DESCRIPTION: SPHERICALL	Y BLUNTED 18° HALF	ANGLE NOSE CO	NE AND FORWARD
PORTION OF CYLINDRICAL SECTION A	PPROXIMATELY 2.7 (ALIBERS IN LEN	GTH.
SCALE = 0.0137		· · · · · · · · · · · · · · · · · · ·	
DRAWING NUMBER: MSFC DWG. 10A00	313 REV. A		
	THEORETIC	<u>AL</u>	ACTUAL MEASURED
DIMENSIONS:	FULL-SCALE	MODEL SCALE	MODEL SCALE
Length	<u>583.9 i</u> n.	<u>8 in.</u>	
Max. Width	<u>146 in.</u>	2 in.	
Max. Depth	<u> 146 in.</u>	2 in.	
Fineness Ratio			
Area			
Max. Cross-Sectional	16,741.5 in. ²	3.142 in. ²	
Planform			
Wetted			
Base	16,497.6 in. ²	3.096 in. ²	

		8	2.160	4.320	62	63	64	65	99	67	89	69	70		
	2	7	1.810	3.620	53	54	55	56	22	58	59	9	61	71*	72*
		9	1.550	3.100	44	45	46	47	48	49	20	51	25		
	L	2	1.377	2.755	35	36	37	38	39	40	41	42	43		
	3	7	1.285	2.570	56	27	28	67	30	31	32	33	34		-
		3	1.007	2.015	11	18	19	20	21	22.	23	24	25		
		2	0.675	1.350	8	6	10	11	12	13	14	15	16		ŀ
	4	1	0.340	0.680	1		2	3	4	5	9	1	7.		
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*Note that 71 and 72 are check orifices located on nose 4 and at longitudinal Station? position.

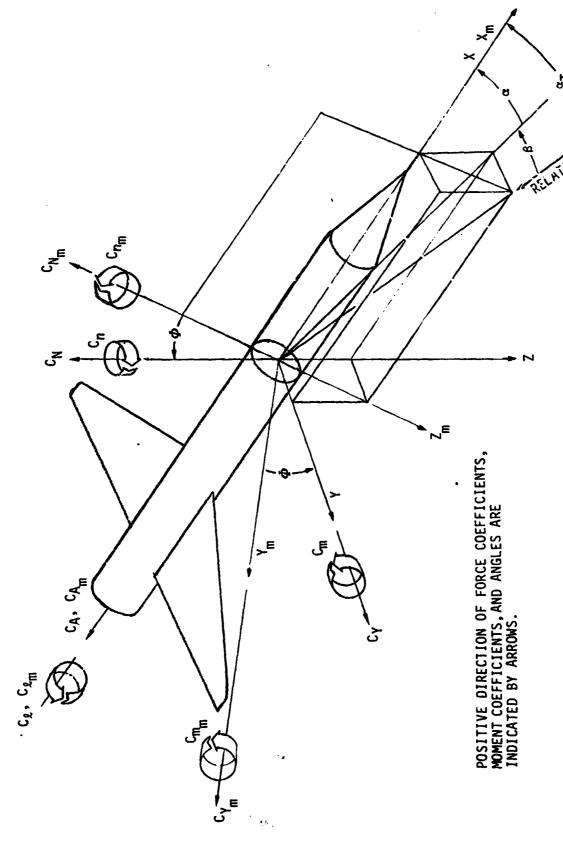
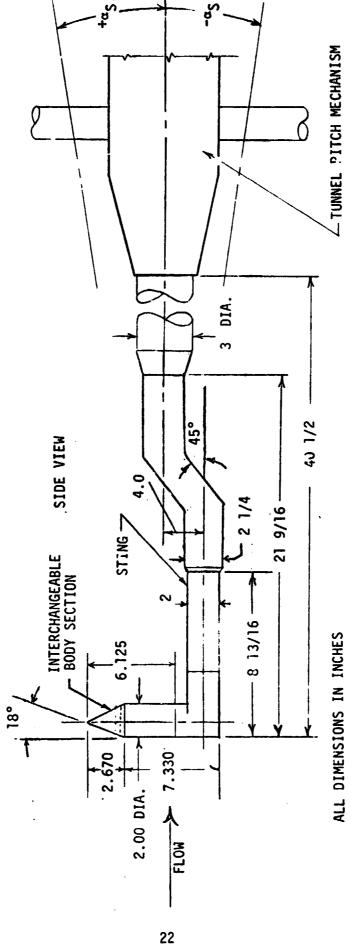


Figure 1. - Body and Missile Axis Systems



Details and Major Dimensions of SRB Forward Section and Sting Figure 2.

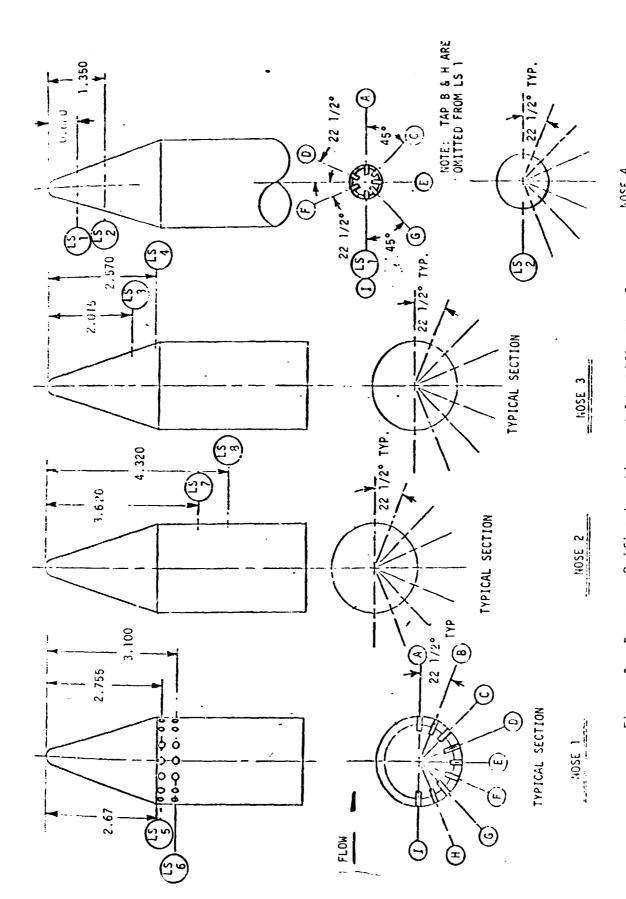


Figure 3. Pressure Orifice Location and Identification System NOSE 4

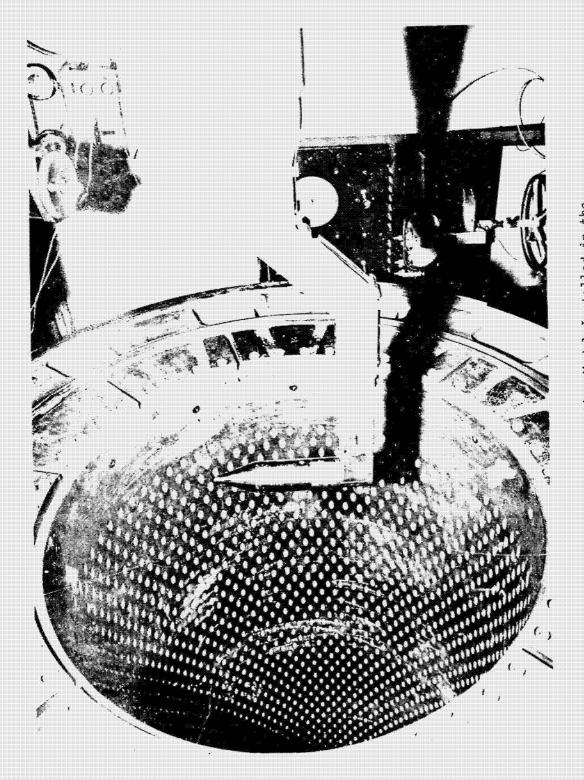
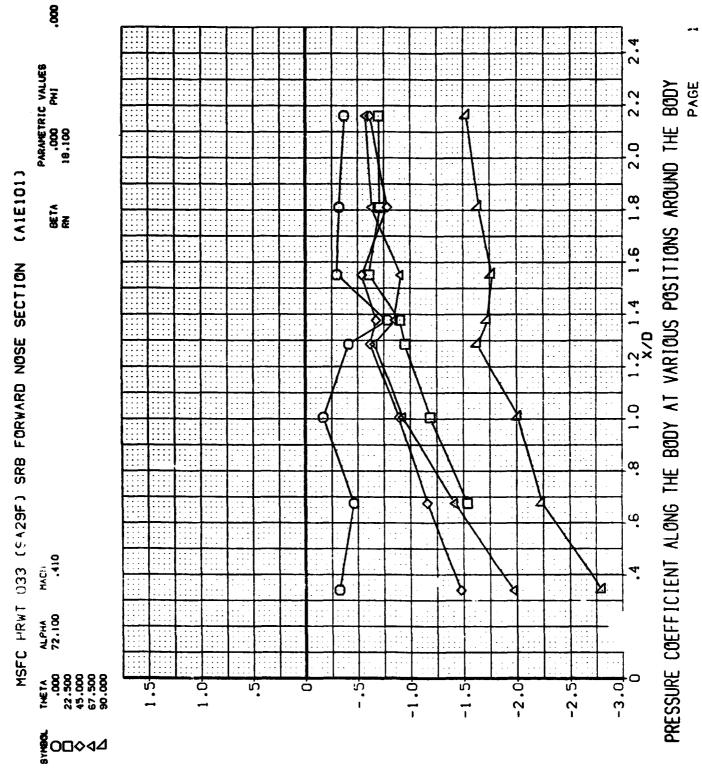
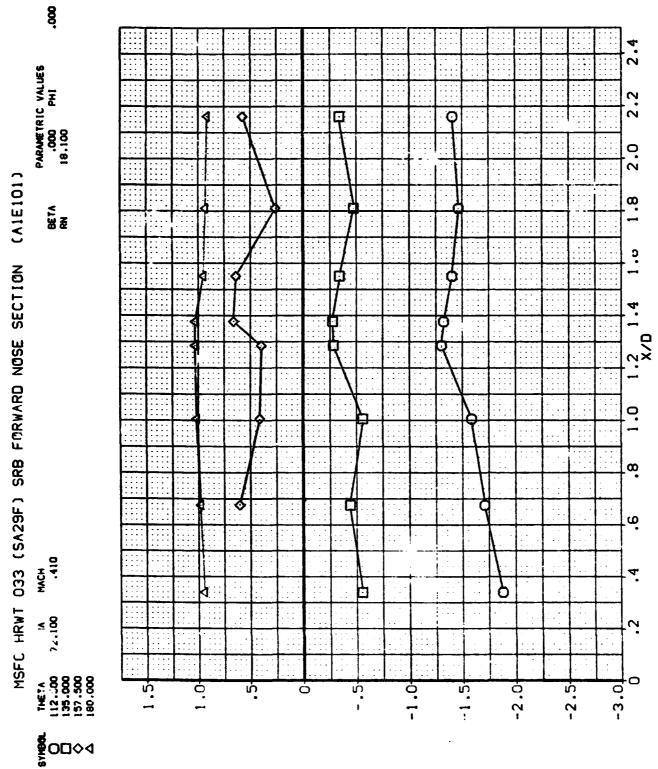


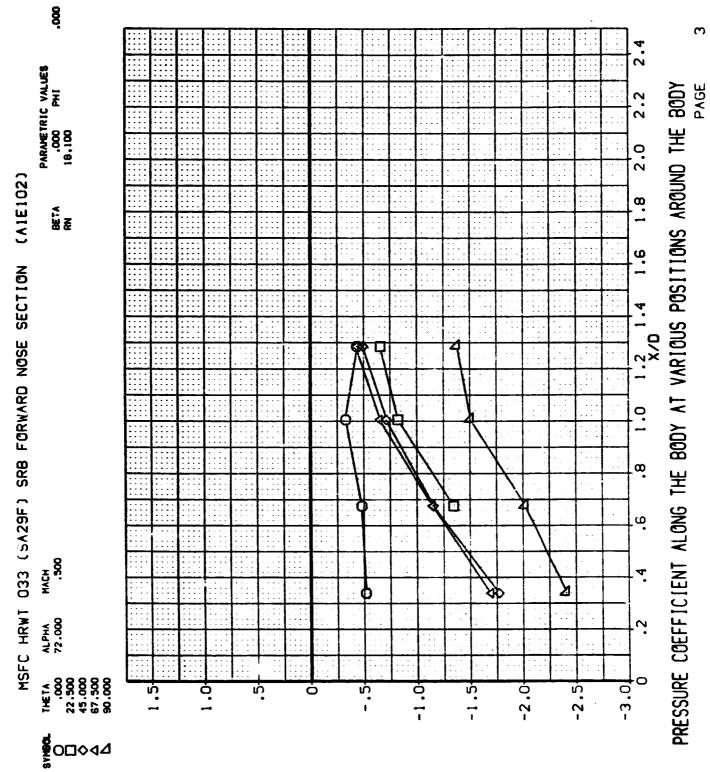
Figure 4. SRB Foresection Model Installed in the MSFC HRWT Transonic Test Section

DATA FIGURES



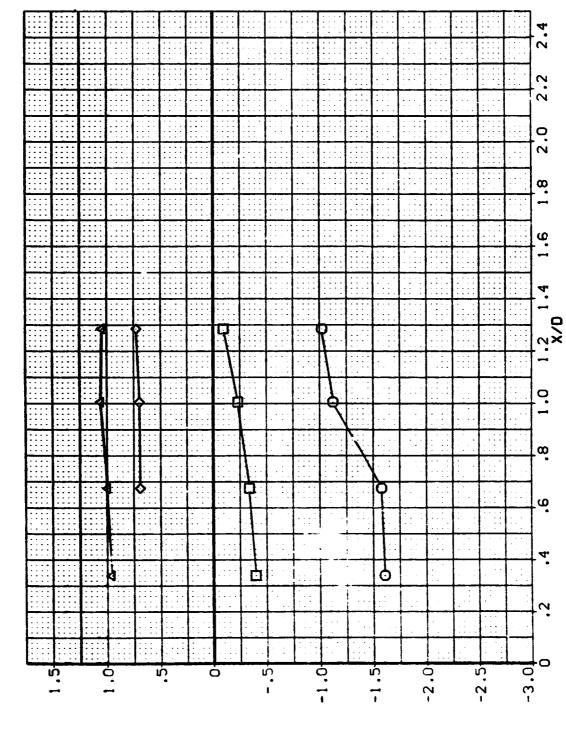


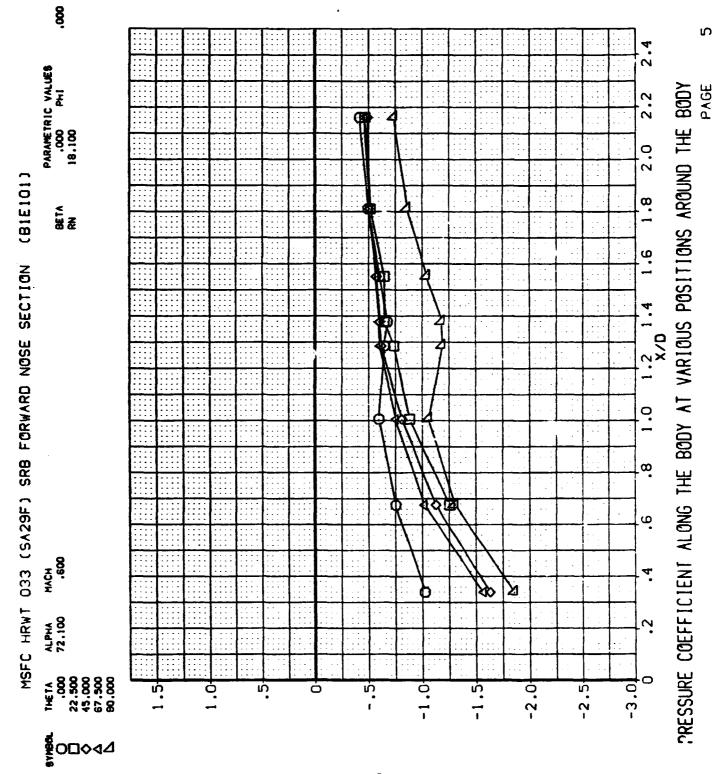
PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE



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ġ MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (A1E102) # ¥ ₹ MACH .500 ALPHA 72.000 THETA 112.300 135.000 157.500 **₹**0□◊4

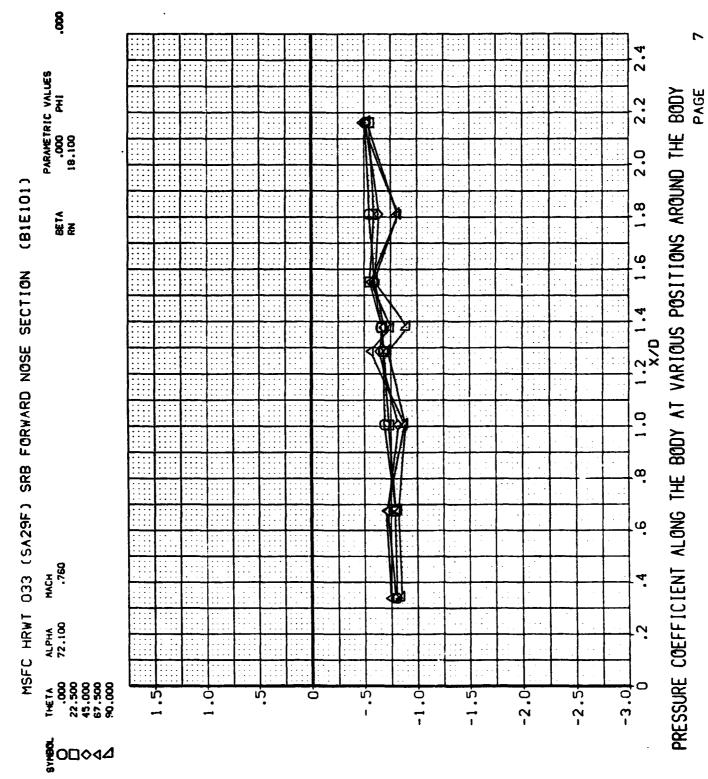




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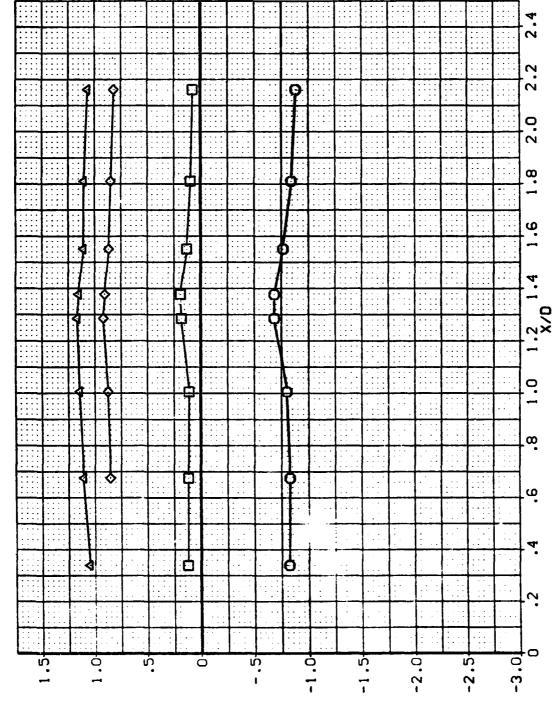


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PARAMETRIC VALUES .000 PHI 18.100 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (BIE101) æ ₹ ₹ ALPHA 72.100 THETA 112.500 135.000 157.500

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PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE

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PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE

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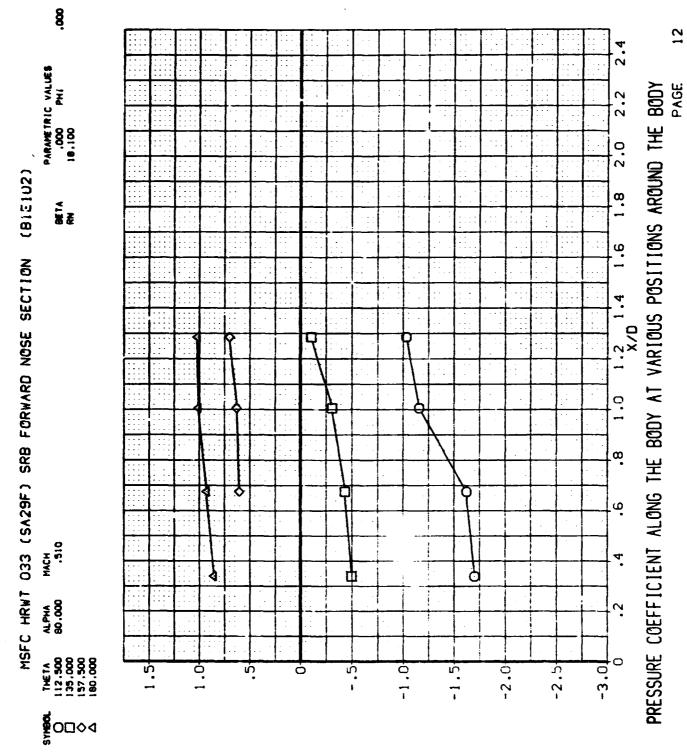
PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE

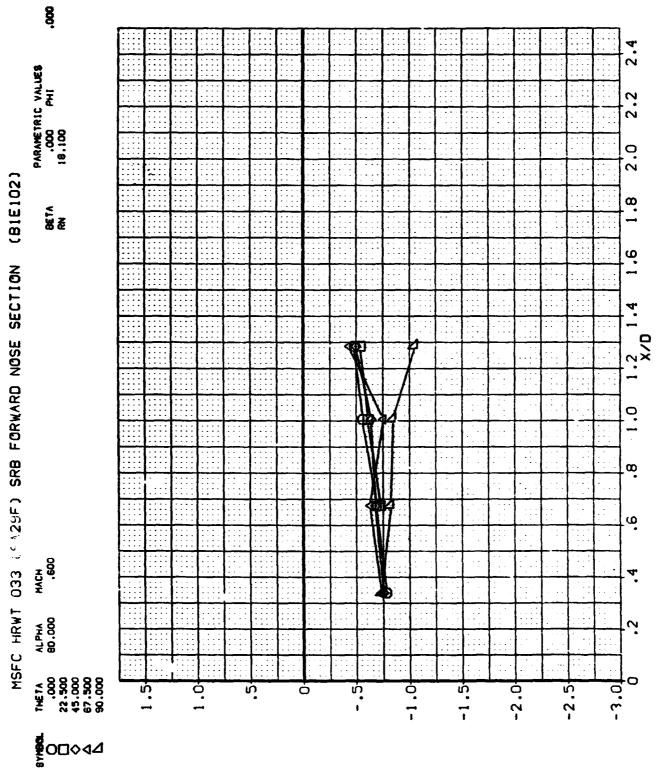
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કું 2.4 2.2 2.0 MSFC HRWT 033 (CA29F) SR. FORWARD NOSE SECTION (BIE102) 8. # ¥ ¥ 9 1.0 œ က MACH .510 ALPHA 80.000 1HETA .000 22.500 45.500 67,500 90.000 9. -1.5 -3.0 1.5 Ś 5. -1.0 -2.0 -2.5 **1** 0□◊4△

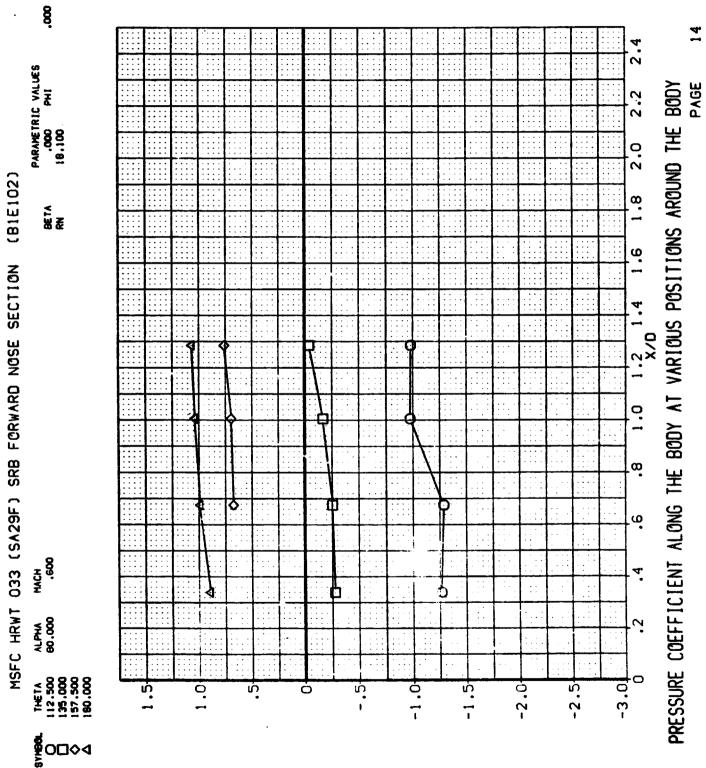
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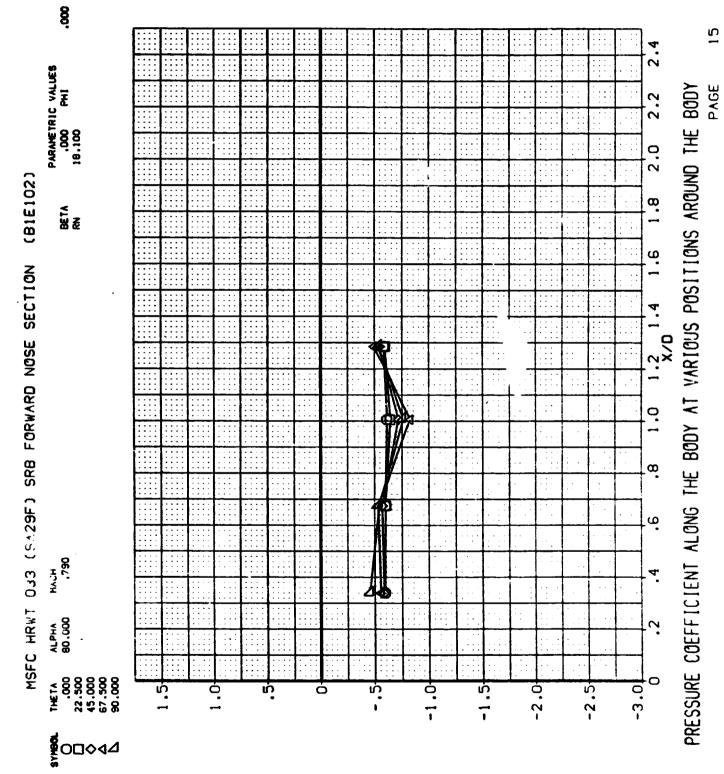


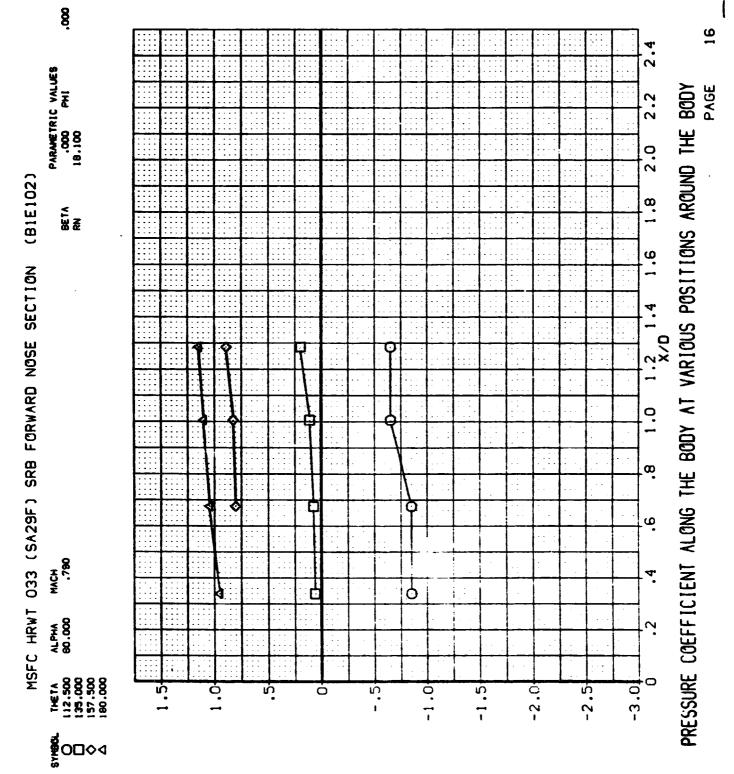


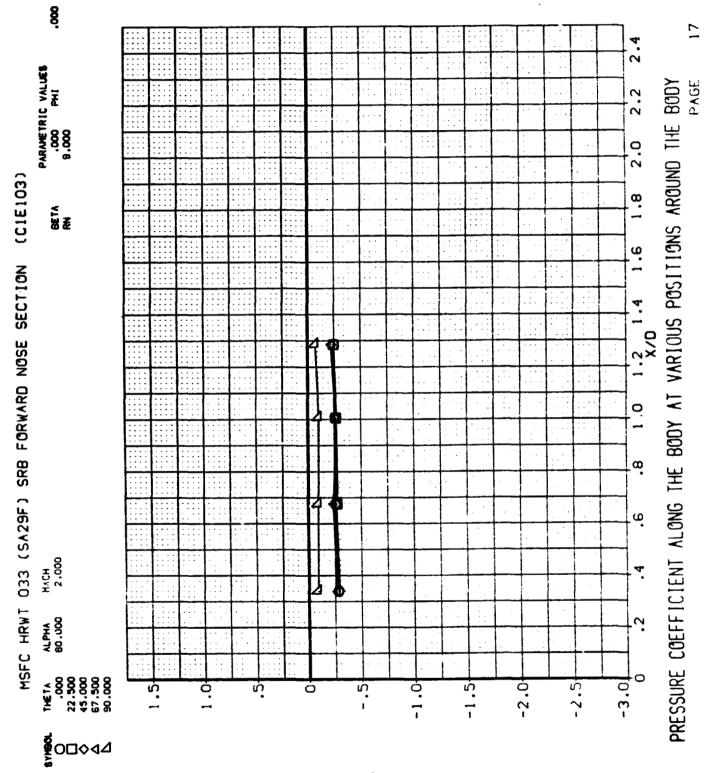
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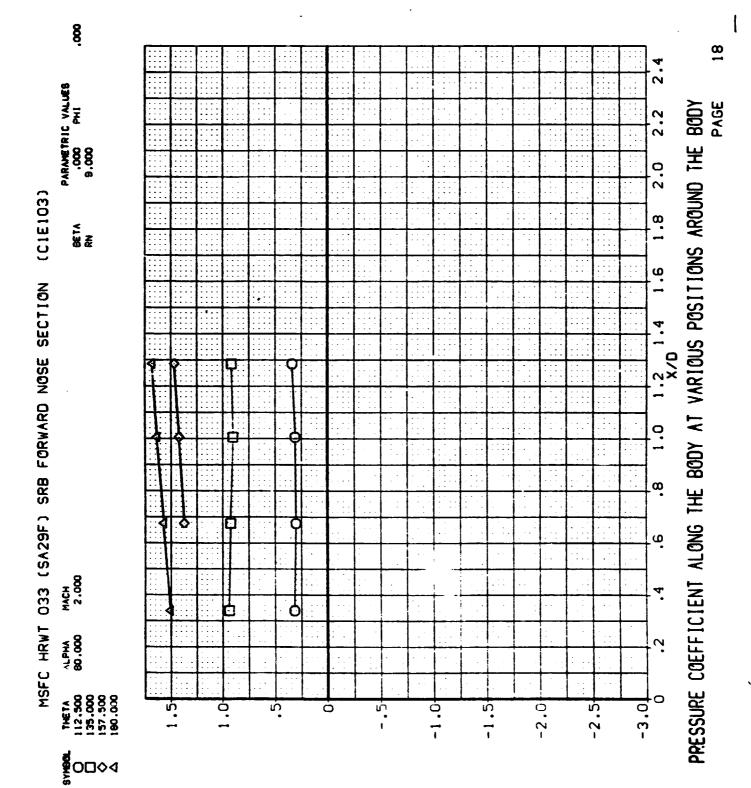


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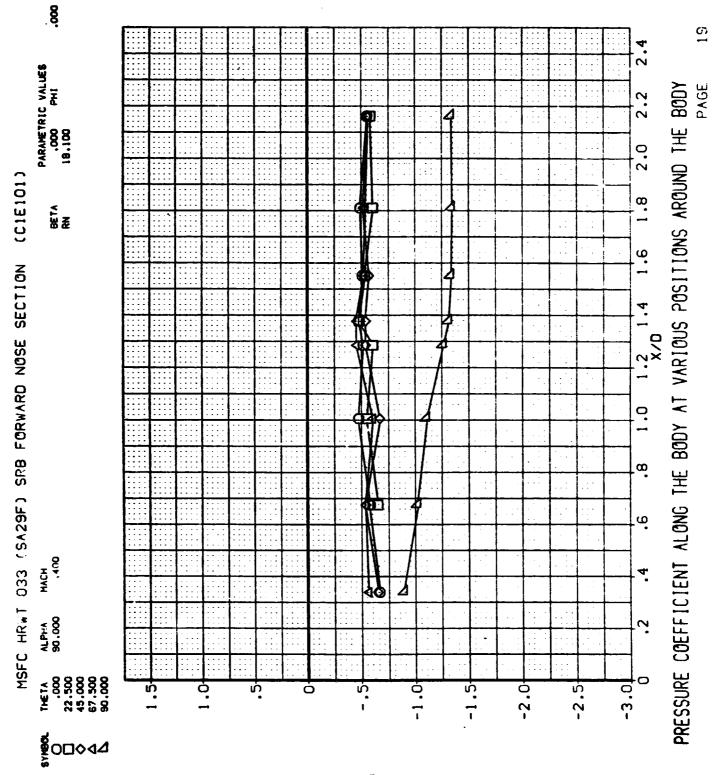


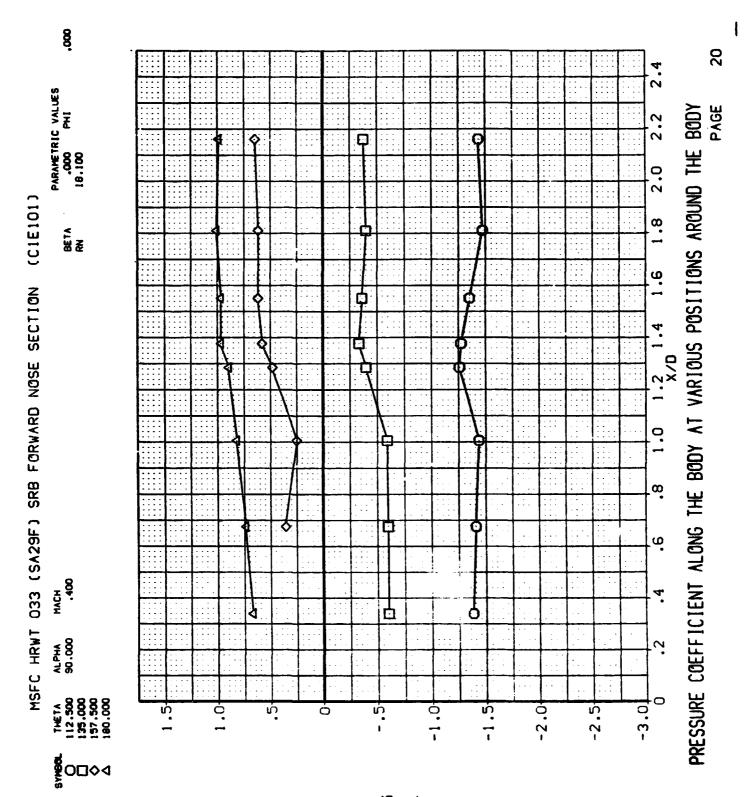




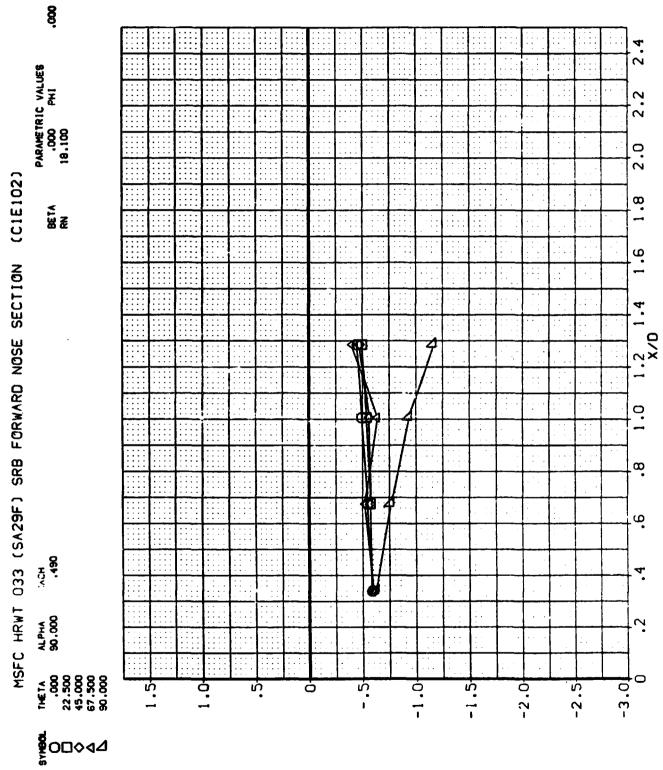








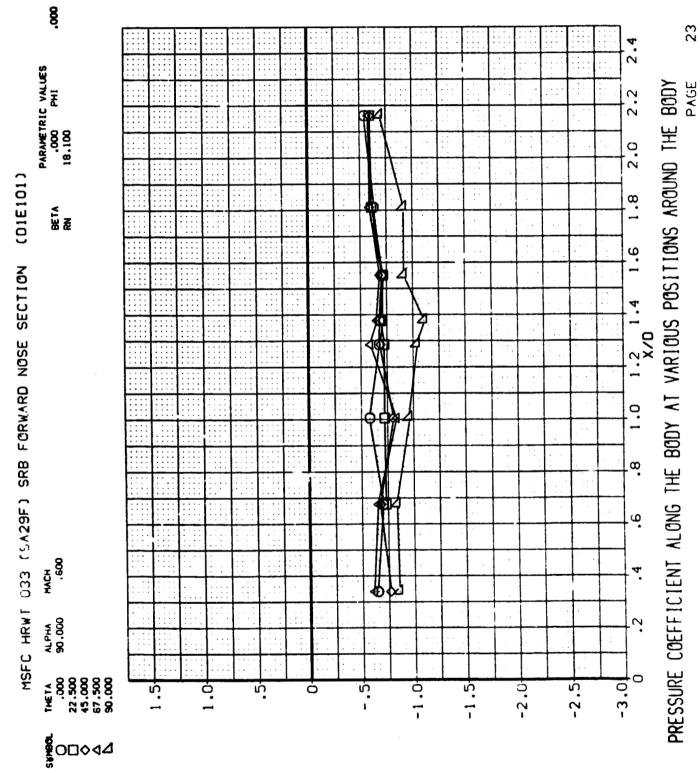




PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE

8 PARAMETRIC VALUES .000 PH1 18.100 2.0 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (CIE102) 8. 86 T 7 X 1.2 1.4 1.6 X/D 0. ω Þ ဖ MACH • 490 .ह ф ALPHA 90.000 THE TA 112.500 135.000 157.500 180.000 -3.0 -1.0 -1.5 -2.0 -2.5 ņ .5 0. Ó § O□◊4

PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE



8 PARAMETRIC VALUES ,000 PHI 18.100 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (DIE101) 88 TX 中 ¢ ₹ 909. 1 7 ALPHA 90.000 THETA 112.500 135.000 157.500 180.000 ņ -1.0 -2.0+ -2.5 5. -1.5 **№**0□◊4

PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE 1.2 1.4 1.6 x/D

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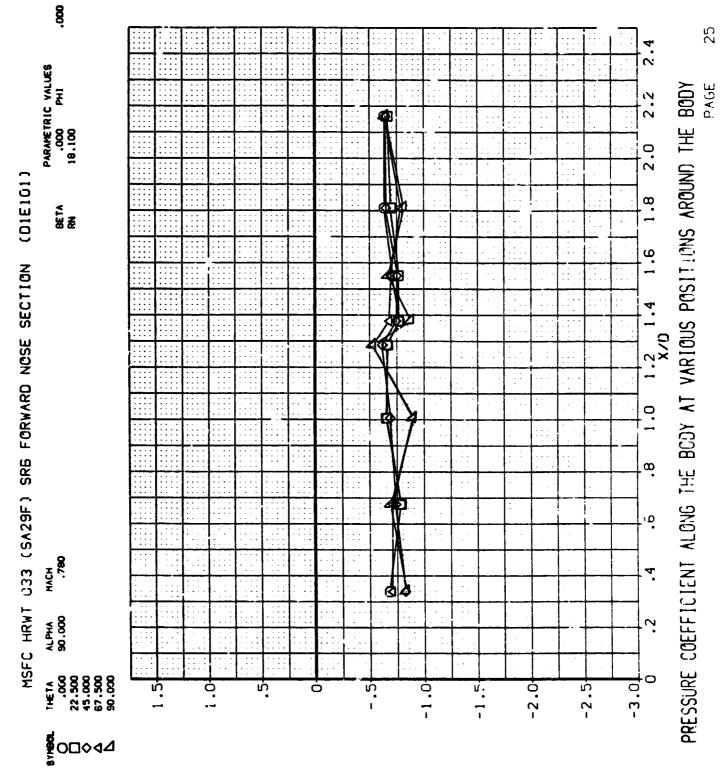
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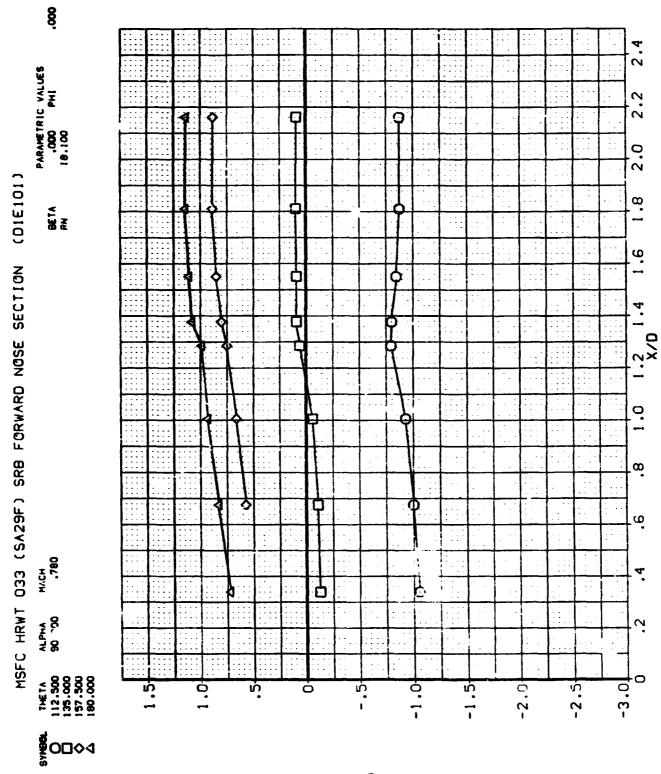
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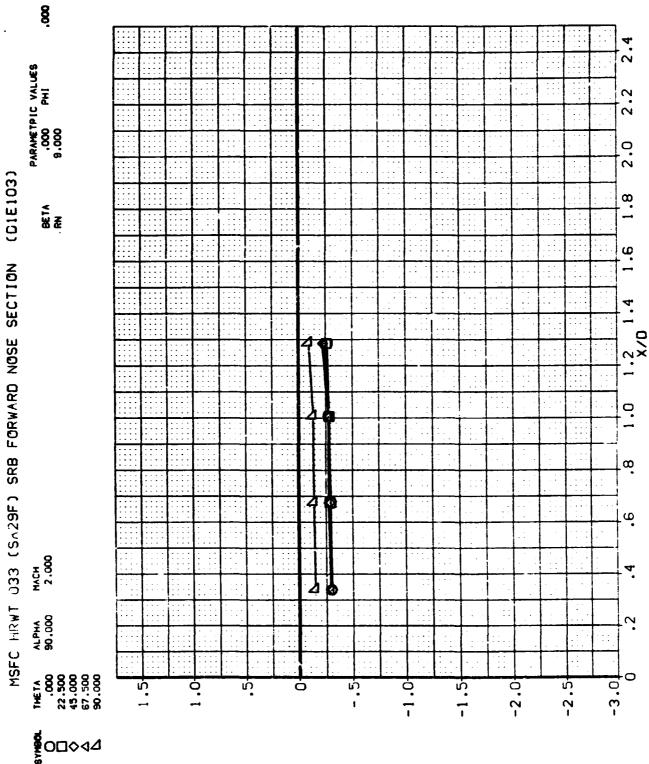
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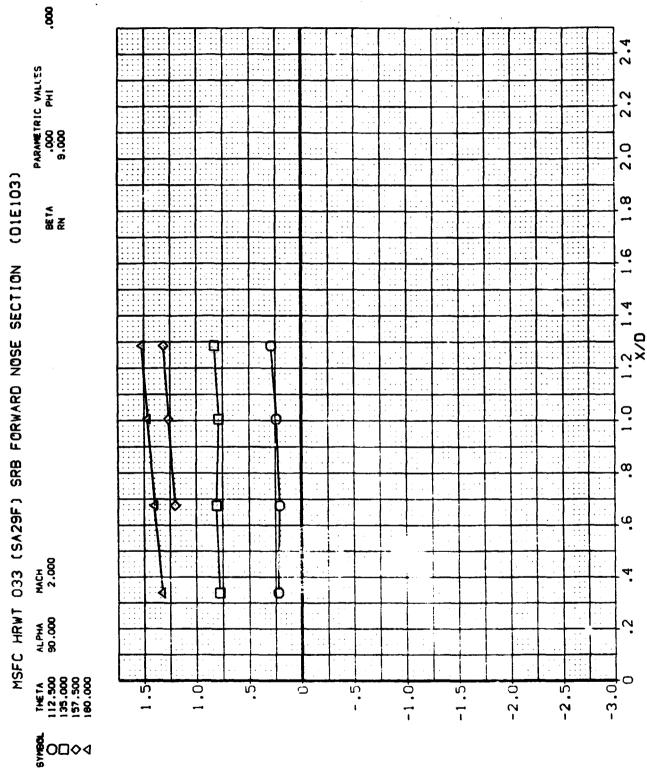


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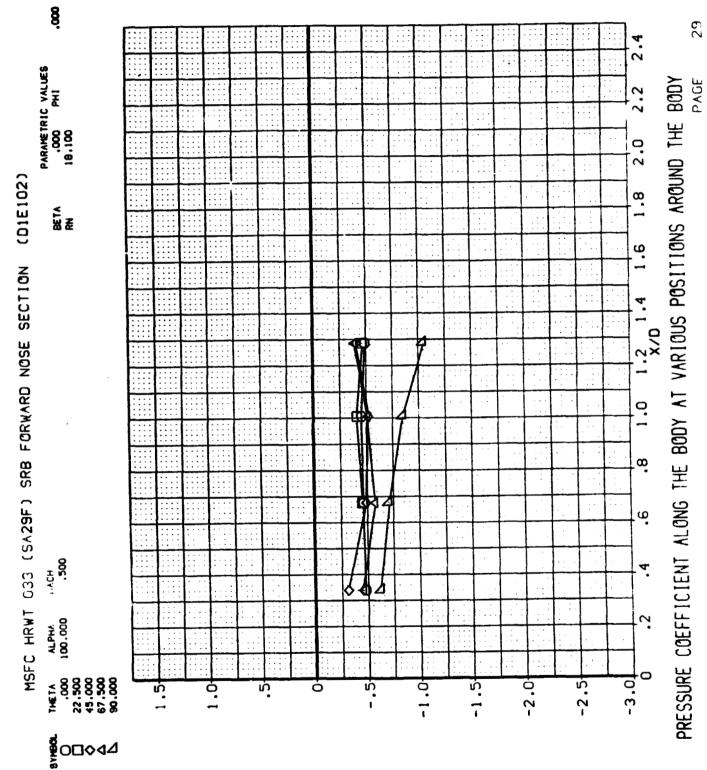
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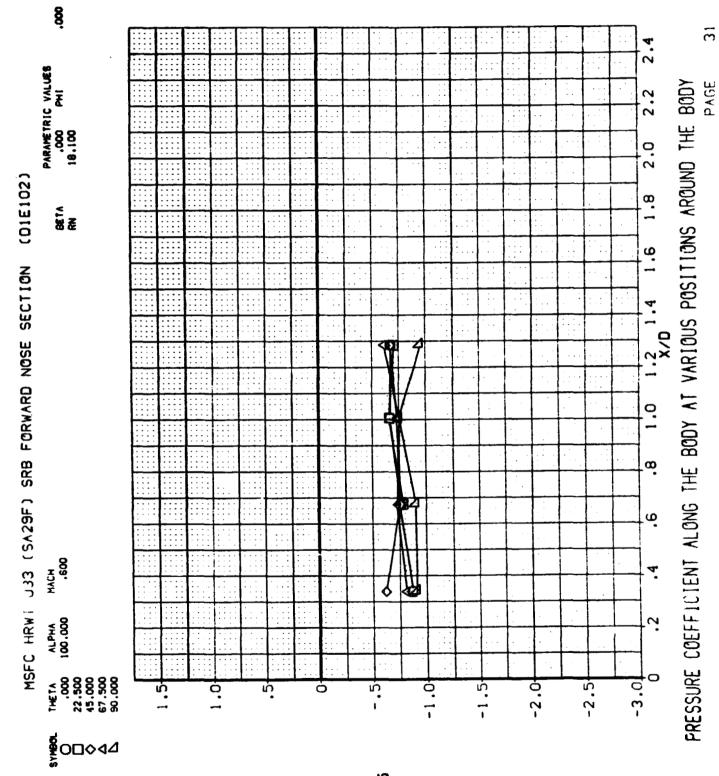
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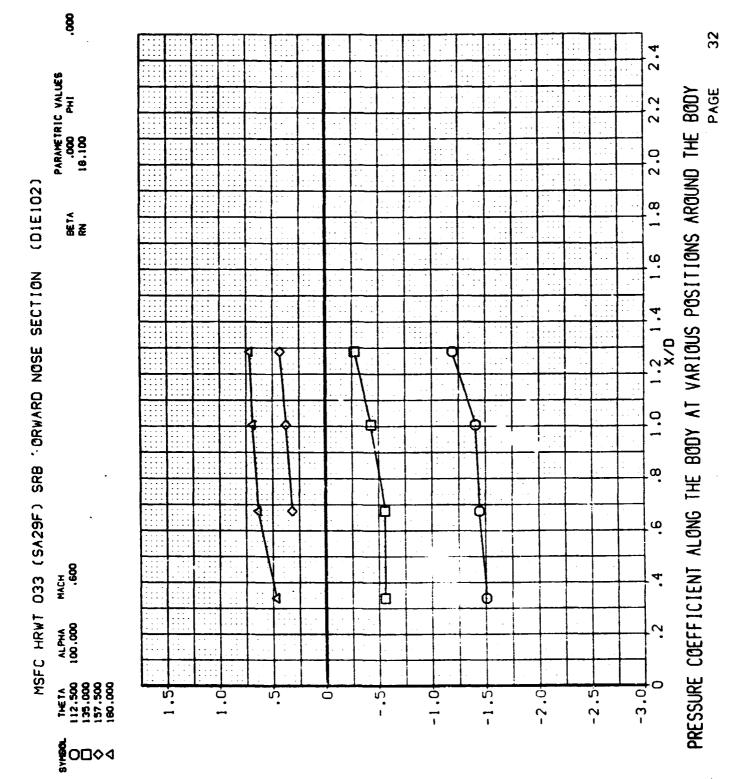
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ģ 2.4 PARAMETRIC VALUES .000 PH: 18.100 2.0 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (DIE102) ... 73 F 1.2 1.4 1.6 X/D 0 ω ဖ ф () ALPHA 100.000 . ! 112.500 135.000 157.500 180.000 -3.0 -2.5 -2.0 0. -1.0 -1.5 5. ņ 5. **№**0□◊⊲

PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE



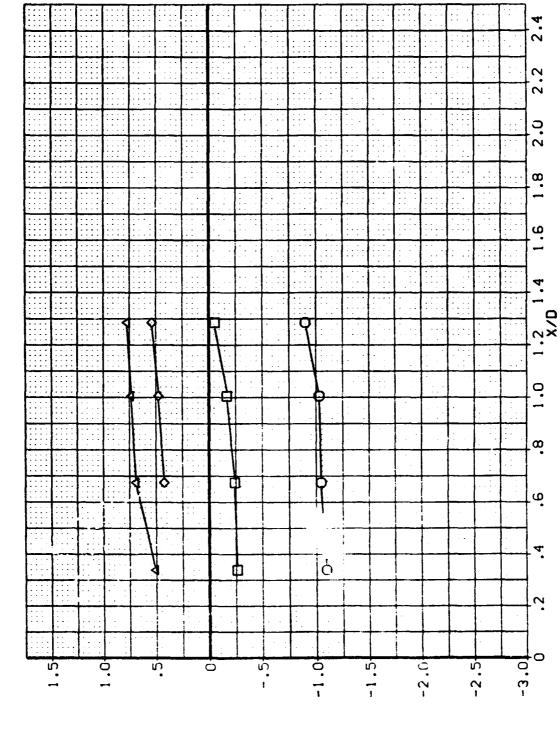


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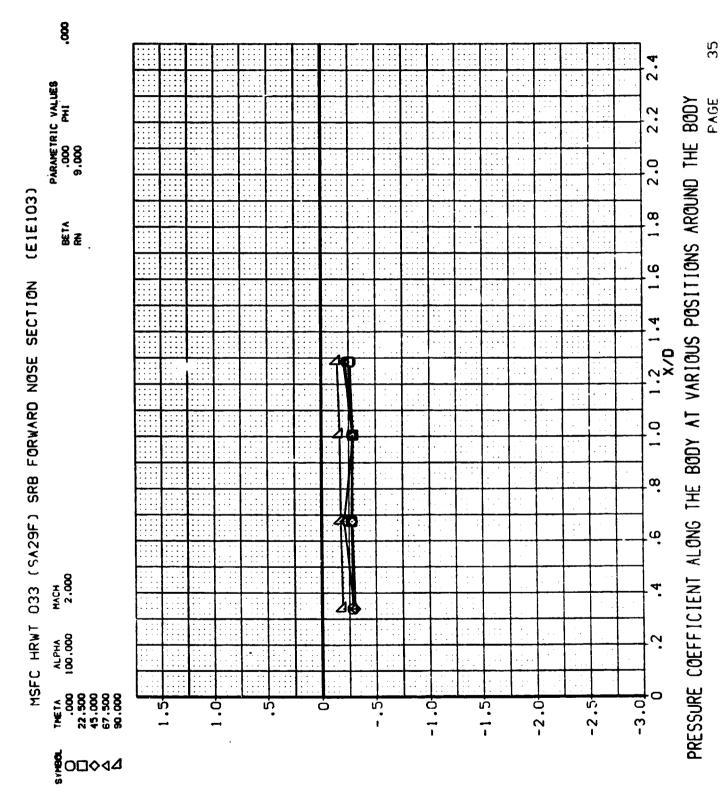
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PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY

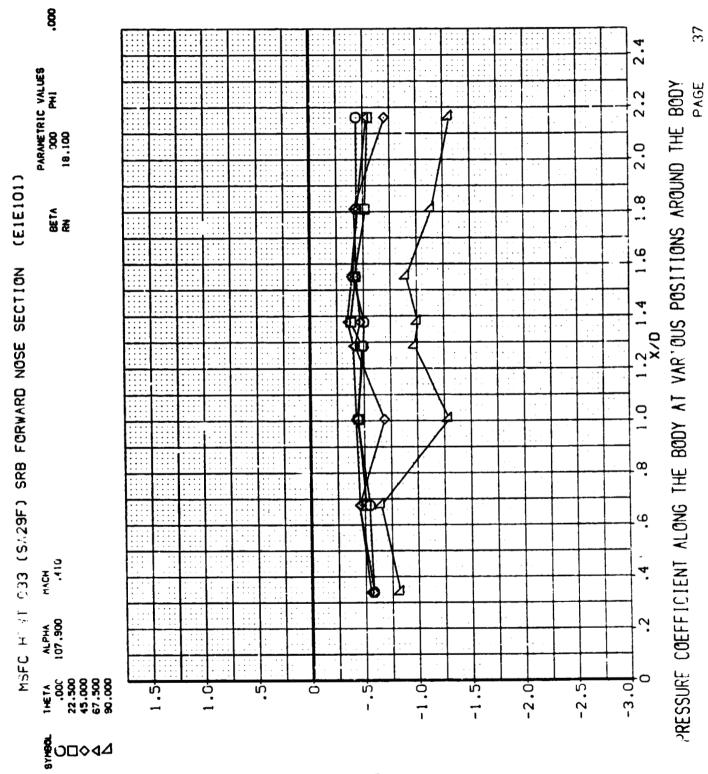
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PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE



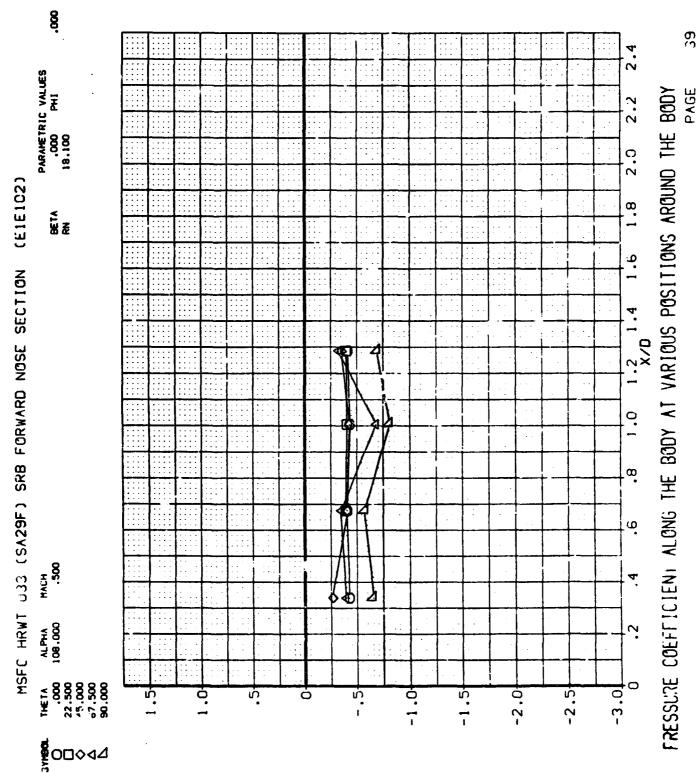
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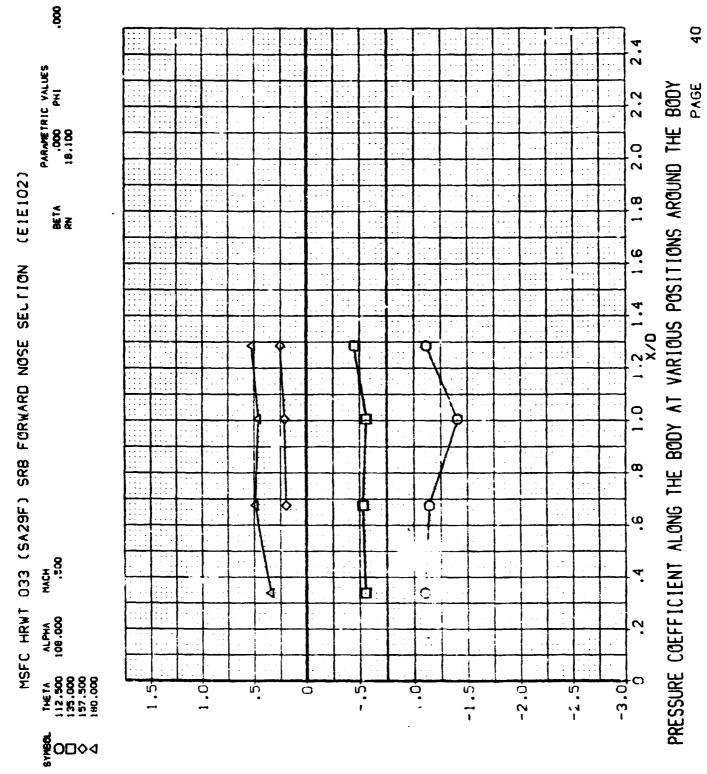
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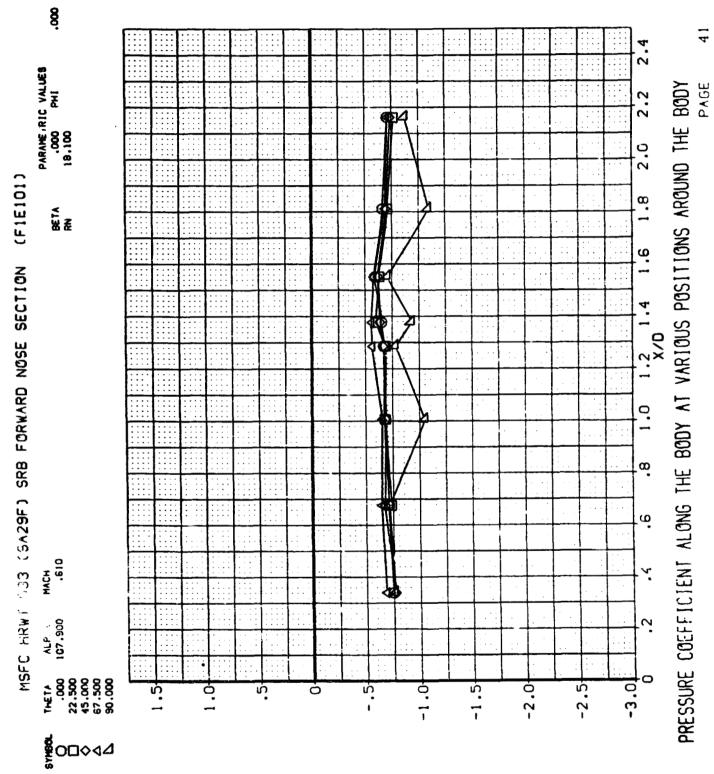
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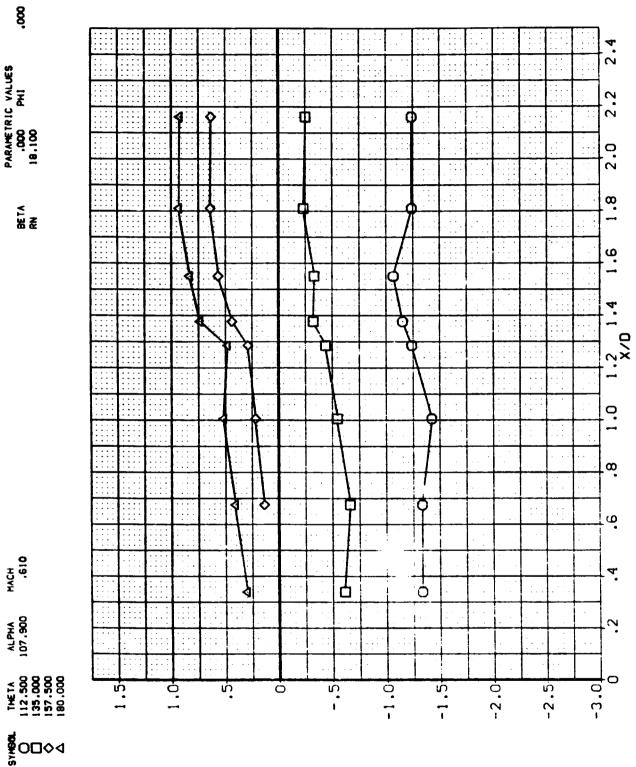
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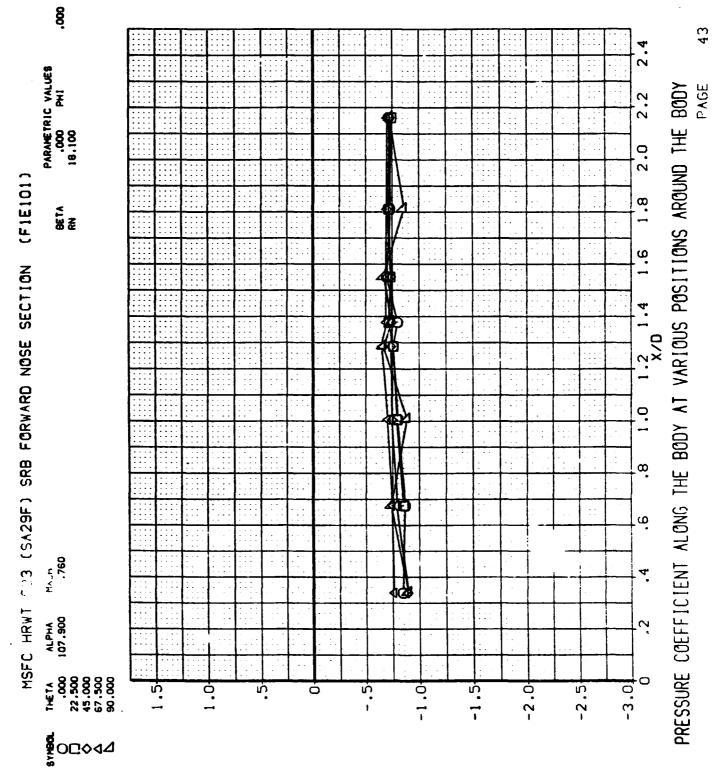


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(F1E101) MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION



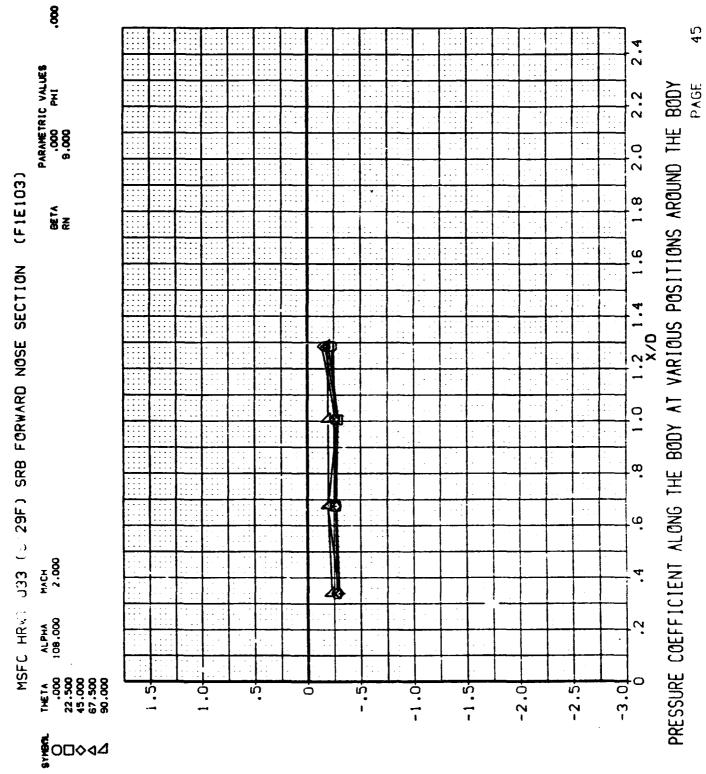
PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE



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ġ 2.4 PARAMETRIC VALUES .000 PHI 18.100 Q 2.0 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (FIE101) 8. ₹ ₹ 0. φ Q ဖ Н d 0 ALPHA 107.900 112.500 135.000 137.500 187.500 -3.0 Ņ -.5 -1.0 -1.5 -2.0 -2.5 5. 0. **№**0□◊4

PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE

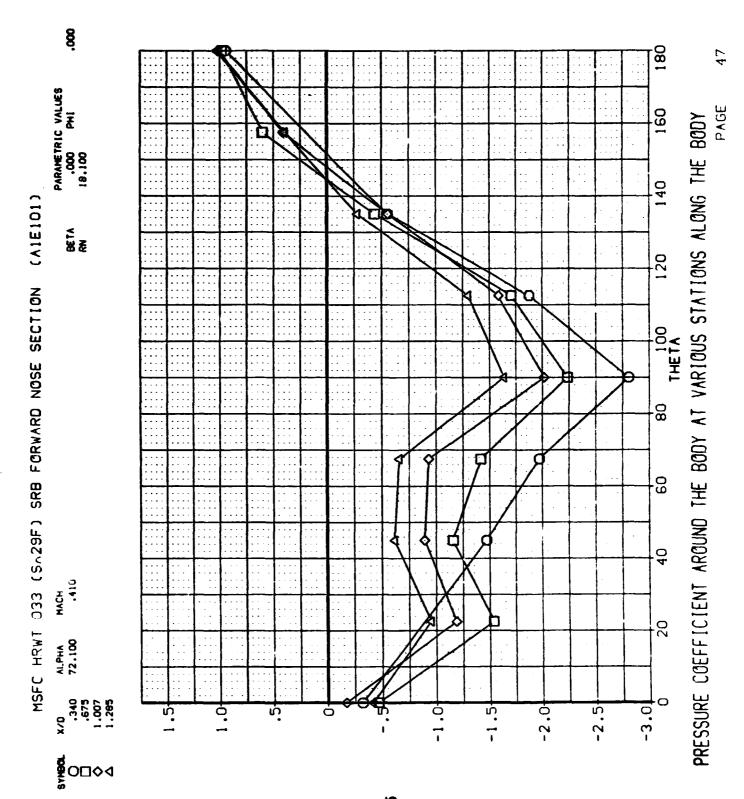


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8 PARAMETRIC VALUES .000 PHI 9.000 2.0 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (F1E103) 8. 8€ T. ₹ 1.6 1.2 1.4 X/D 0. МАСН 2.000 Ь AL PHA 108.000 7KETA 112.500 135.000 157.500 -3.0 -2.5 -1.0+ -2.0 5 5. -1.5 **₹**0□◊4

PRESSURE COEFFICIENT ALONG THE BODY AT VARIOUS POSITIONS AROUND THE BODY PAGE

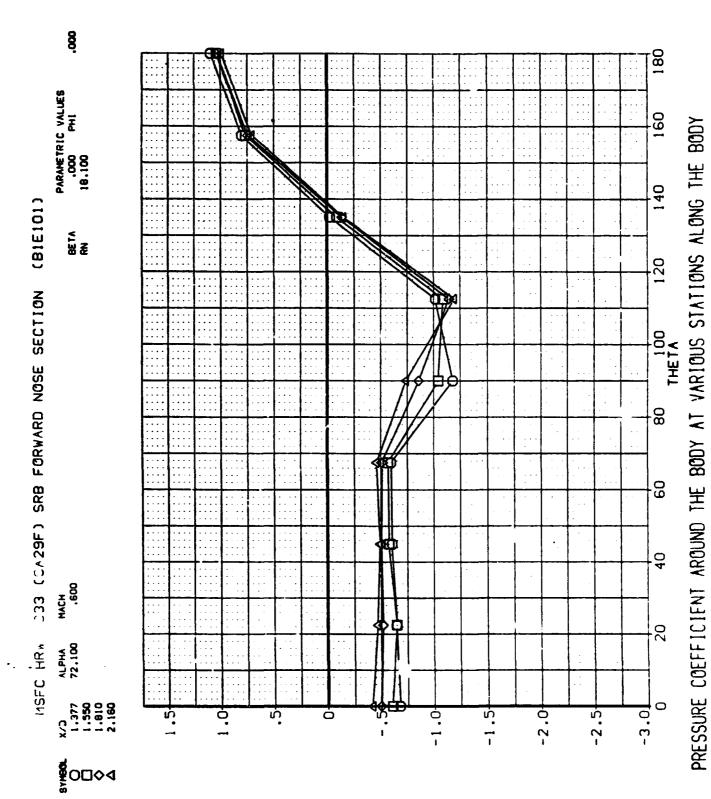
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8 180 48 PARAMETRIC VALUES .000 PHI PRESSURE COEFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY PAGE 140 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (A1E101) 8 TA 7 X 120 100 THE TA 80 <u>6</u>0 40 ALPHA 72.100 -3.0 -2.0 -2.5 .5 -1.0 -1.5 Ó **ਛੂ**0□◊4

8 180 PARAMETRIC VALUES .000 PHI 18.100 160 PRESSURE COEFFICIENT ARGUND THE BODY AT VARIOUS STATIONS ALONG THE BODY 140 MSFC HPWT 033 (SA29F) SRB FORWARD NOSE SECTION (A1E102) 중 도 100 THETA 80 口 60 40 MACH .500 20 ALPHA 72.000 ±0.€--2.5 -1.5 1.5 <u>o</u>. ניו -1.0 -2.0 ò **8** O□◊4

8 180 20 PARAMETRIC VALUES .000 PHI 18.100 PAGE 160 PRESSURE COEFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY 140 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (BIE101) ₹ ₹ 100 THETA 80 60 4 ALPHA 72.100 -1.5 -2.0 -2.5 -3.0| 1.5 Ŋ 5. Ó **№**0□◊4

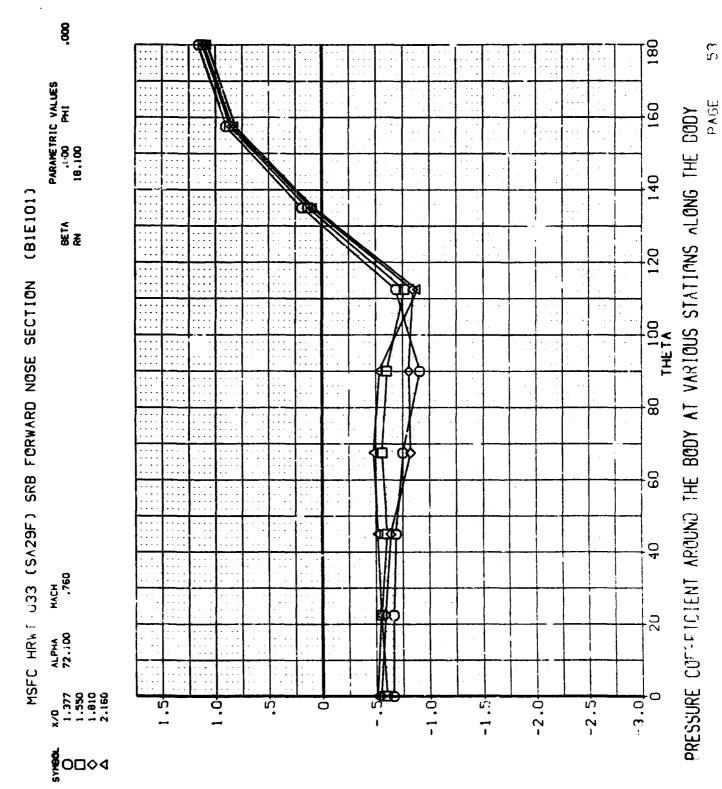


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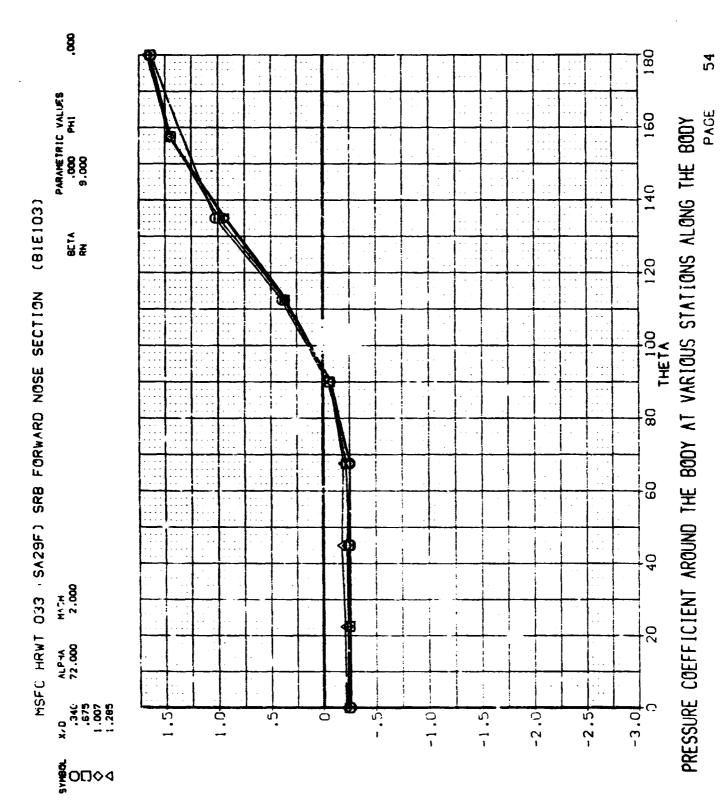
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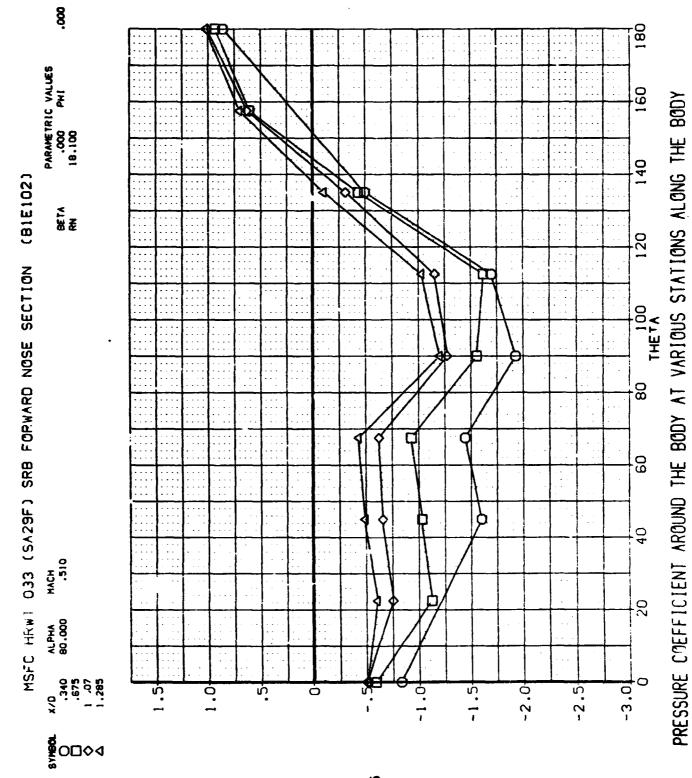
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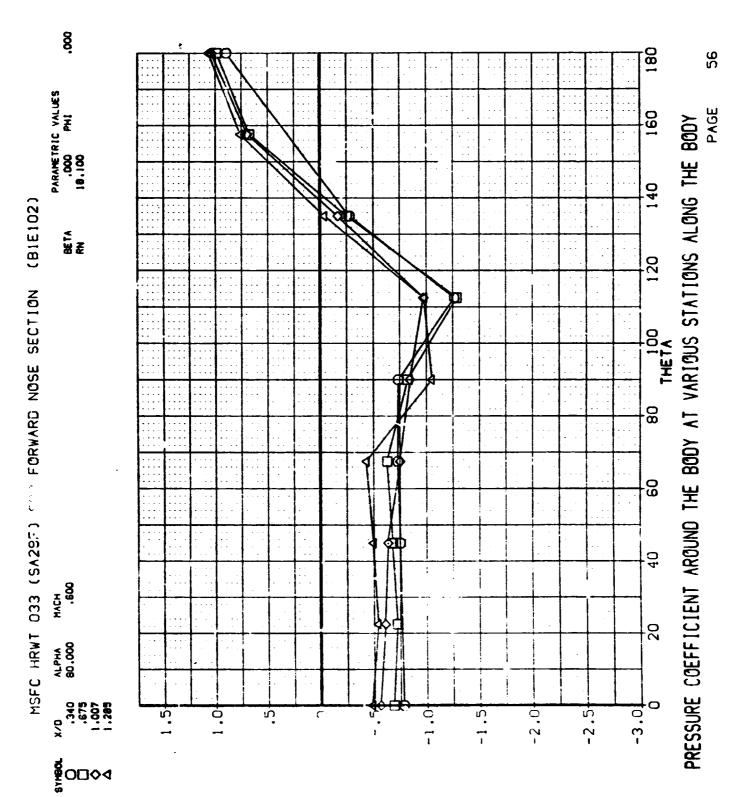


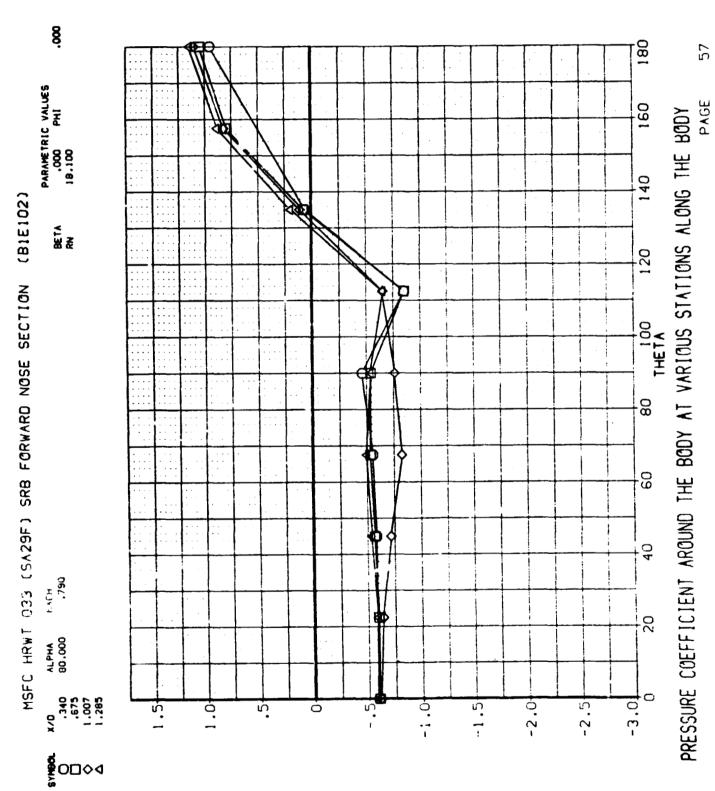
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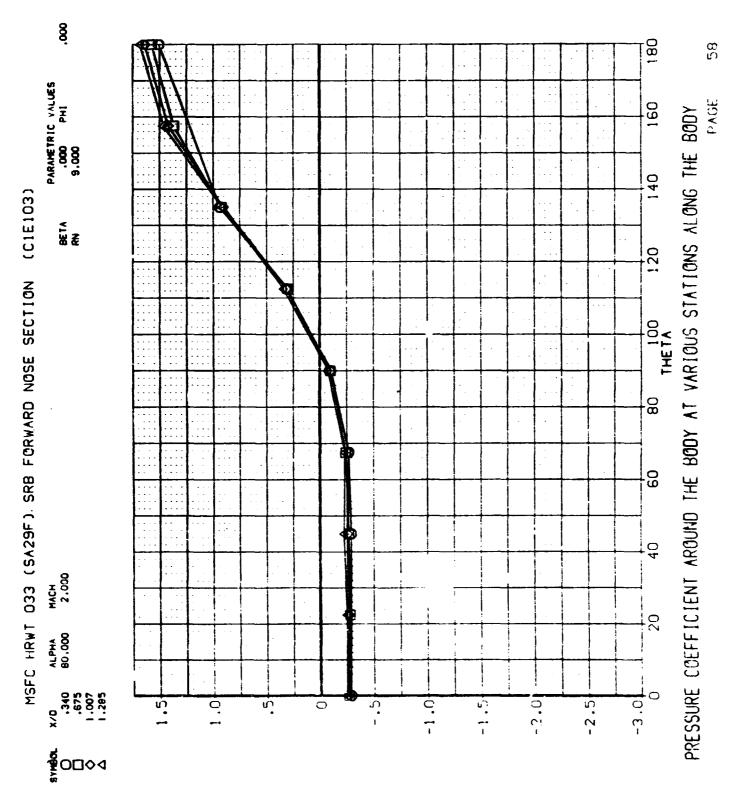


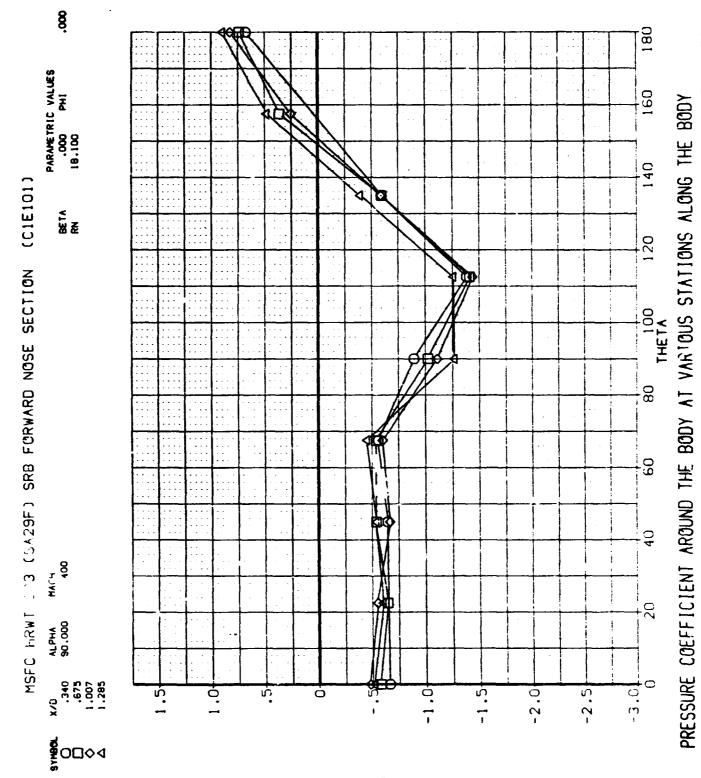
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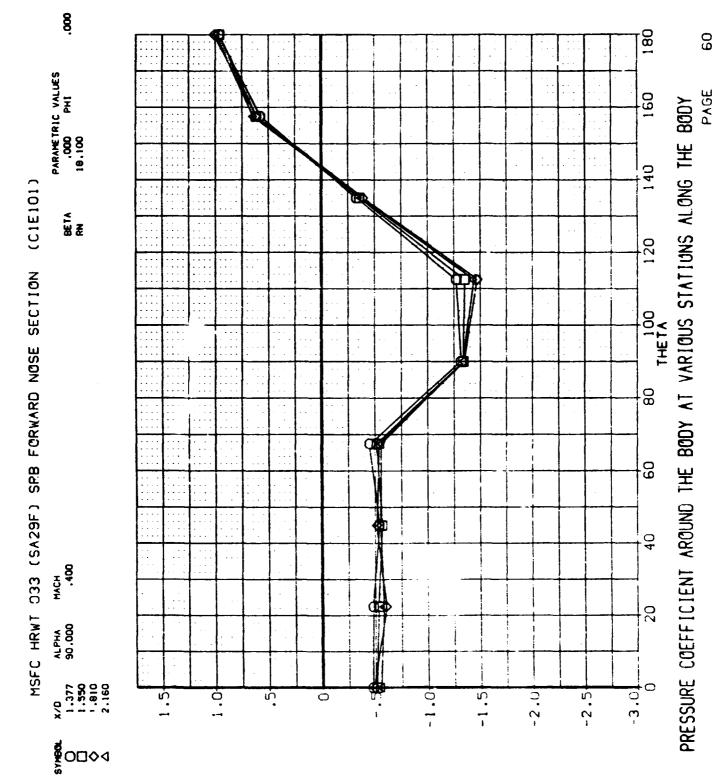




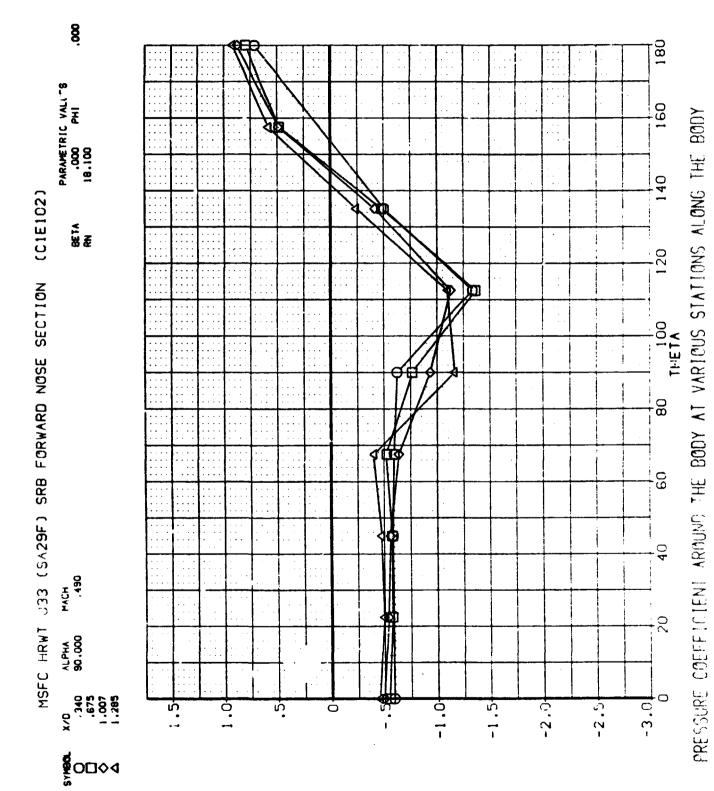








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8 180 PARAMETRIC VALUES .000 341 18.100 160 140 MSFC HRWI 033 (SA29F) SRB FORWARD NOSE SECTION (DIE101) 78 YE 7 100 THETA 80 60 40 3ACH .600 20 AL PHA 90.000 x/D .340 .675 1.007 -3.0L -.5 -2.5 1.0 3. -1.0+ -2.0 -1.g **№**0□◊4

PAGE PRESSURE COEFFICIENT AROUND THE EODY AT VARIOUS STATIONS ALONG THE BOD?

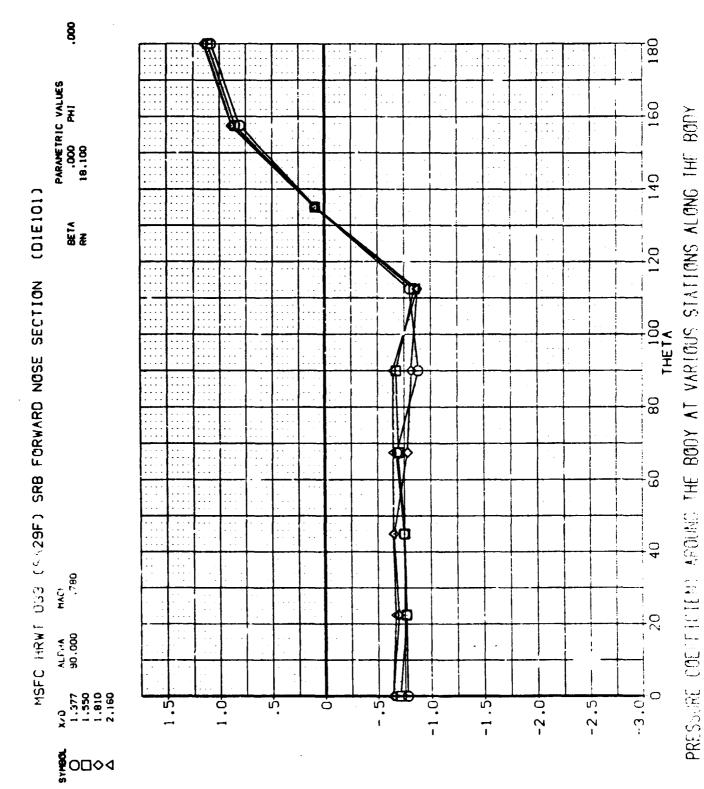
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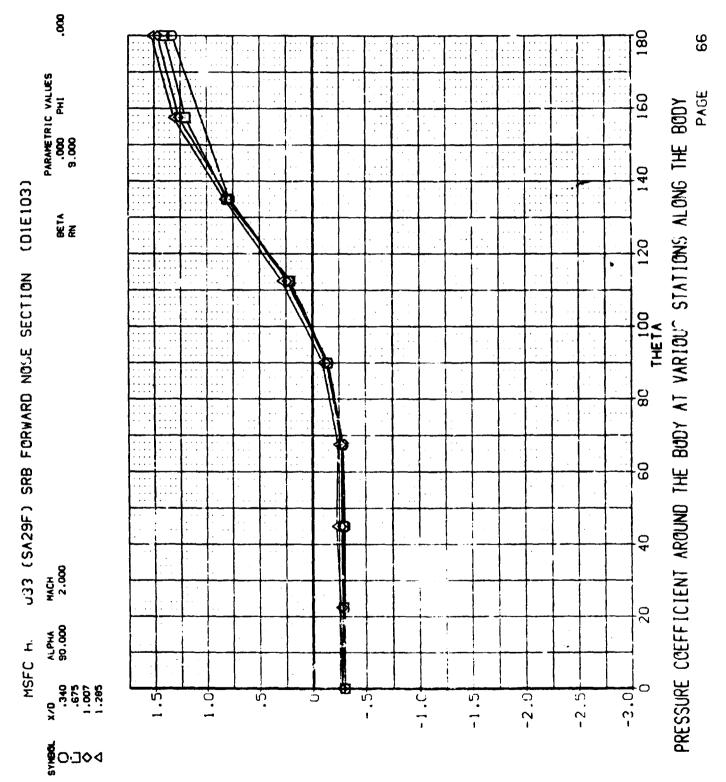
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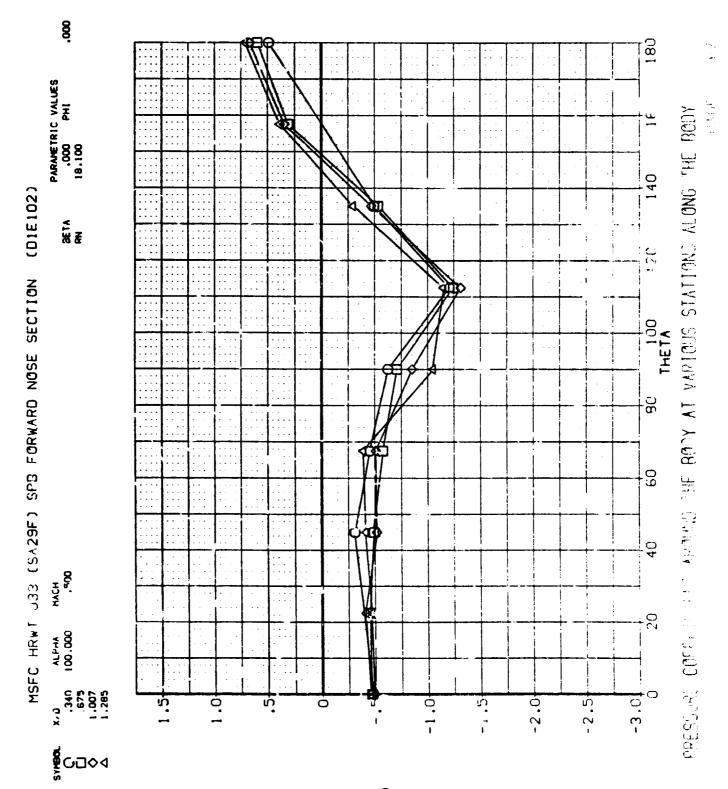
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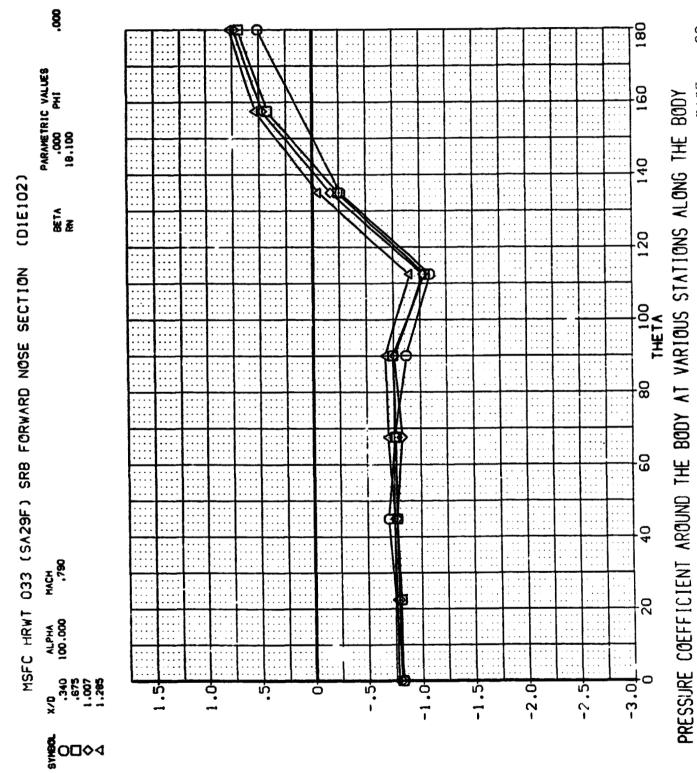
8 180 64 PARAMETRIC VALUES .000 PHI 18.100 PAGE 160 PRESSURE COEFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (DIE101) ₹ ¥ 120 100 THETA 80 9 ALPHA 90.000 x/D .340 .675 1.007 1.5 <u>ن</u> -2.0+ -3.0 6. Ŋ. -1.5 -2.5 -1.0 Ò **№**0□◊◊





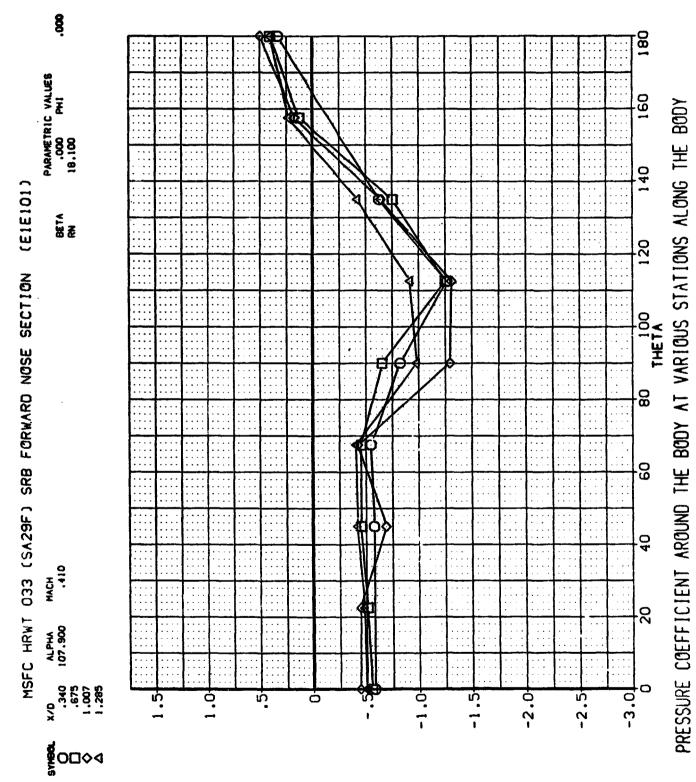


<u>.</u> 180 89 PARAMETRIC VALUES .000 PHI 18.100 160 PRESSURE COEFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (DIE102) # ₹ 7 -1.5 -2.5 -3.0| 'n -2.0 **2**0□◊⊲



ġ 180 PARAMETRIC VALUES .000 PHI 9.000 PRESSURE COEFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY PAGE 160 140 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (E1E103) # % * THETA MACH 2.000 ALPHA 100.000 -2.5 -3.0 -1.5 -2.0 Ę, **№**0□◊4

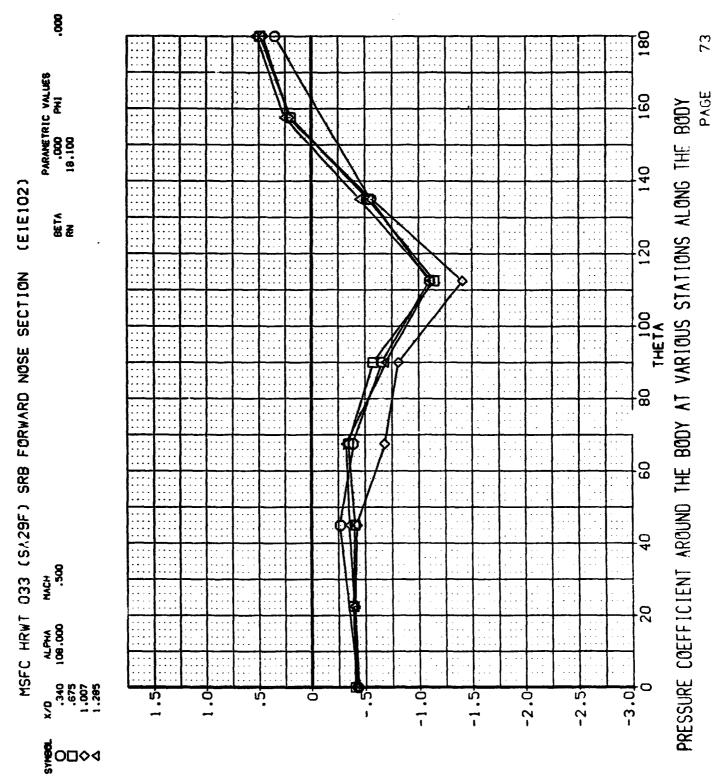
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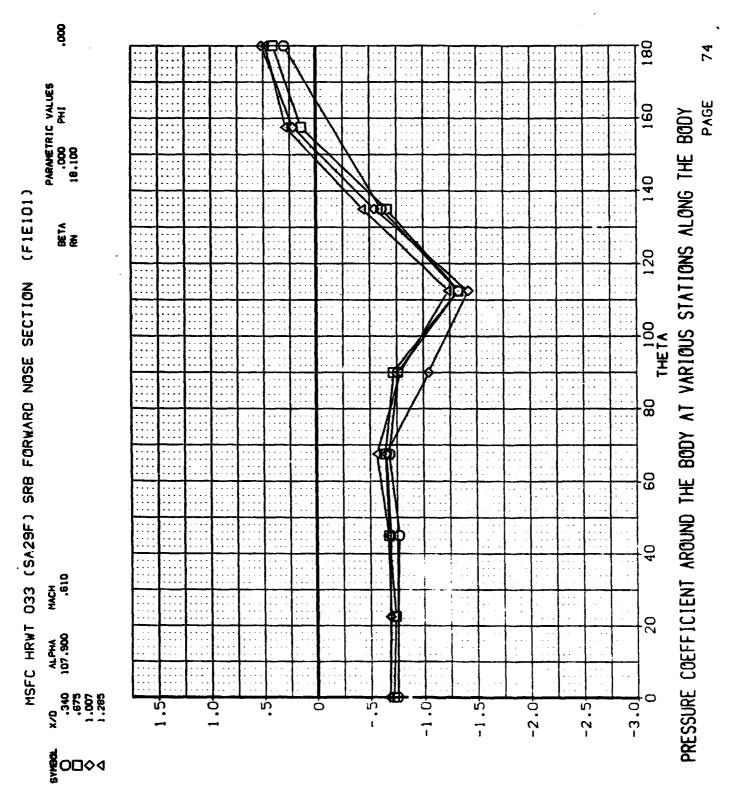


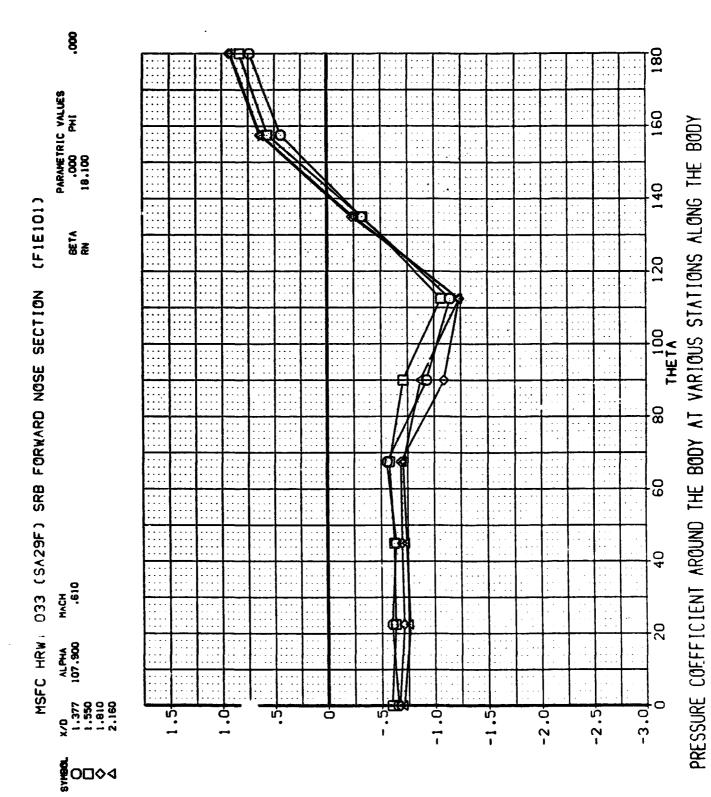
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ģ 180 PARAMETRIC VALUES .000 PHI 18.100 160 140 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (E1E101) ## ## ₹ THETA 9 40 50 ALPHA 107.900 -3.0 -1.0 -1.5 -2.0 -2.5 0.1 5 **₹**0□◊4

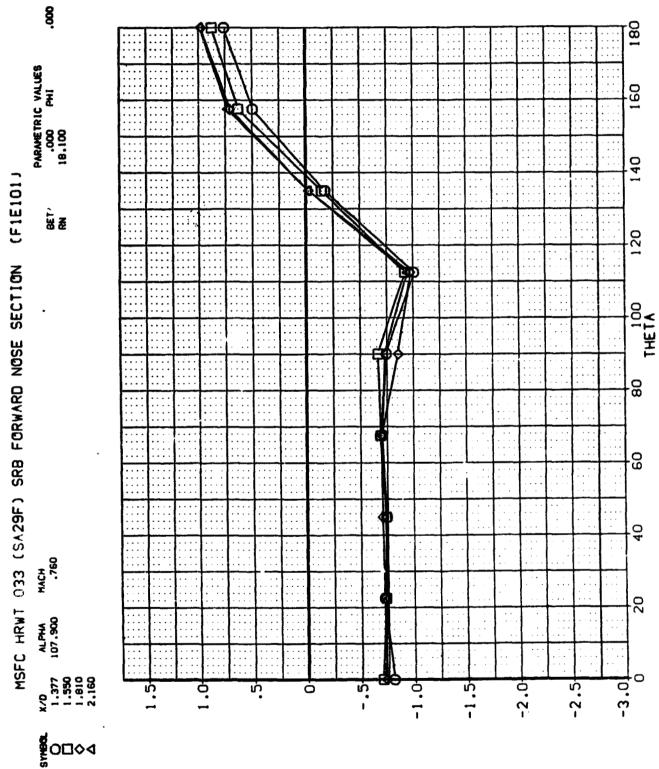
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8 180 9/ PARAMETRIC VALUES .000 PHI 18.100 PAGE 160 PRESSURE COEFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY 140 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (FIE101) # # ¥ ¥ 120 100 THETA 90 40 <u>5</u>0 ALPHA 107.900 -2.5 -3.0 -1.5 -2.0 'n **§**O□◊∢



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PRESSURE CREFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY

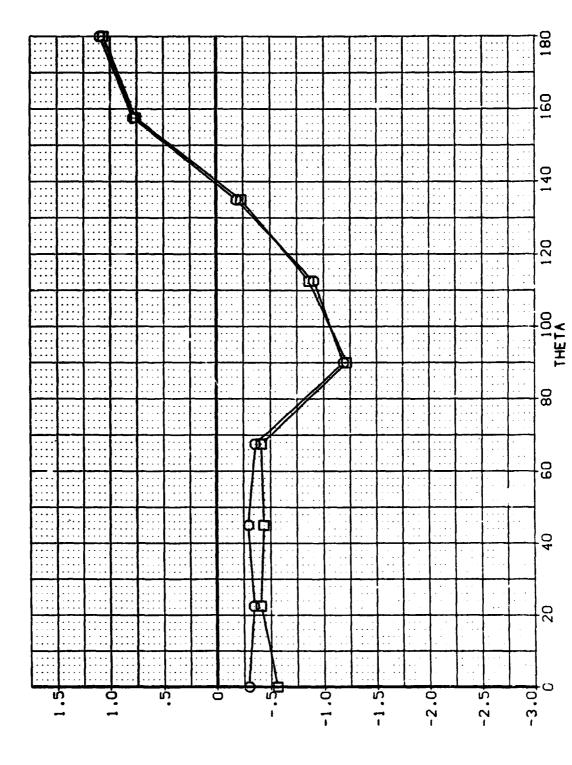
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8 180 78 PARAMETRIC VALUES ,000 PHI 9,000 PAGE 160 PRESSURE COEFFICIENT AROUND THE BODY AT VARIOUS STATIONS ALONG THE BODY 140 MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION (F1E103) 88 T X 120 100 THETA 90 40 MACH 2.000 ALPHA 108.000 ±0.€--1.0 -1.5 -2.0 -2.5 Ş . 6 0□◊4

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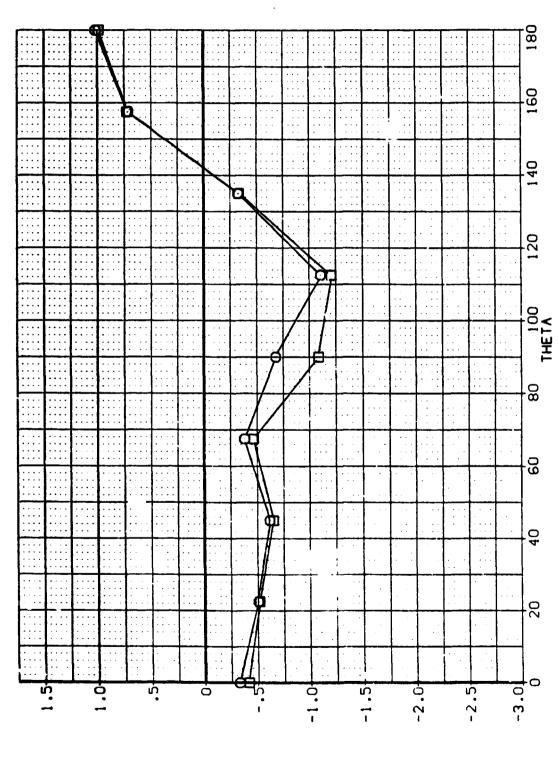
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180,000

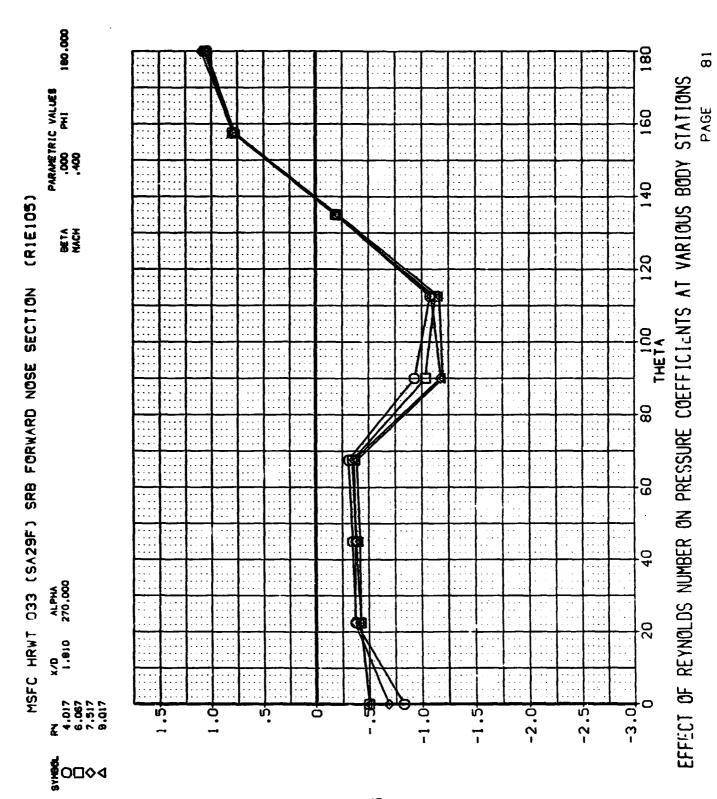


EFFECT OF REYNOLDS NUMBER ON PRESSURE COEFFICIENTS AT VARIOUS BODY STATIONS

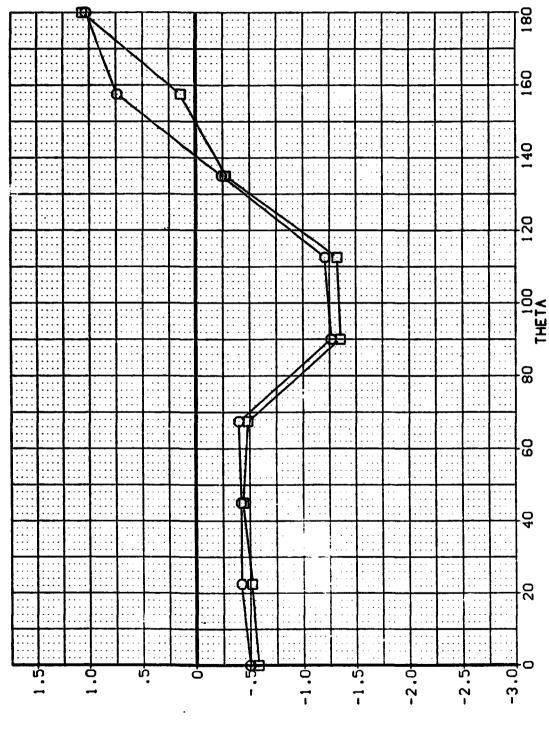
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EFFECT OF REYNOLDS NUMBER ON PRESSURE COEFFICIENTS AT VARIOUS BODY STATIONS PAGE

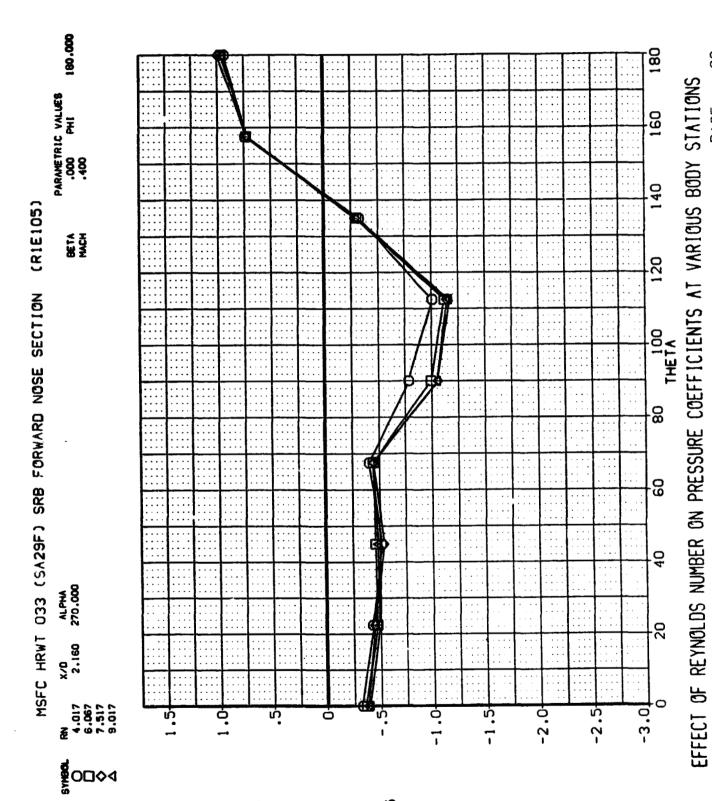


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EFFECT OF REYNOLDS NUMBER ON PRESSURE COEFFICIENTS AT VARIOUS BODY STATIONS PAGE

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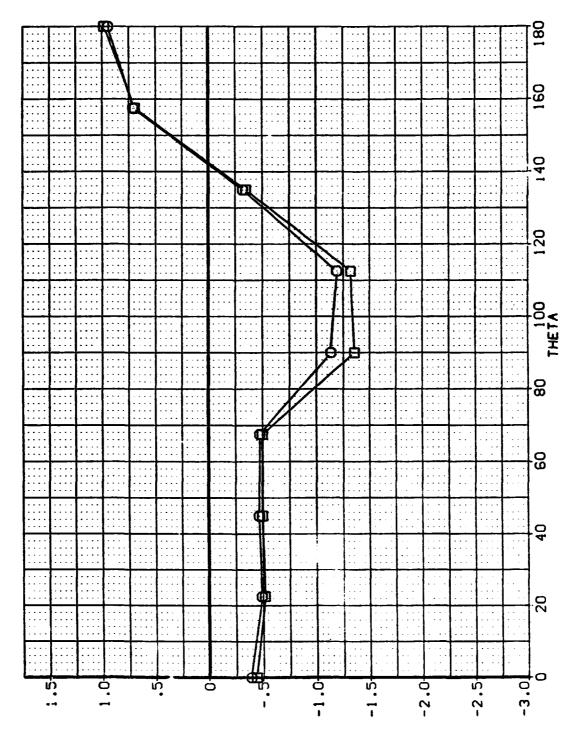


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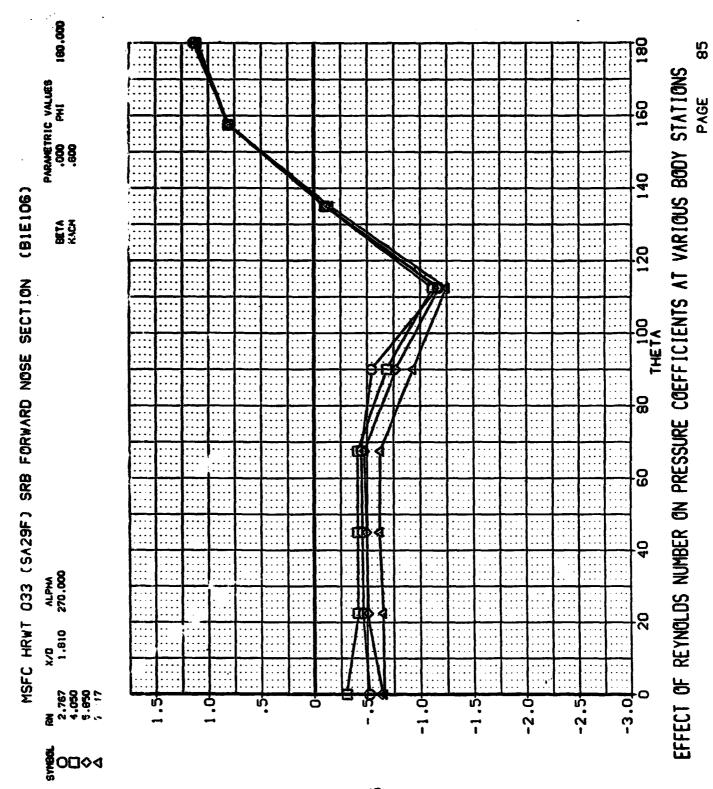
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180.000



EFFECT OF REYNOLDS NUMBER ON PRESSURE COEFFICIENTS AT VARIOUS BODY STATIONS PAGE

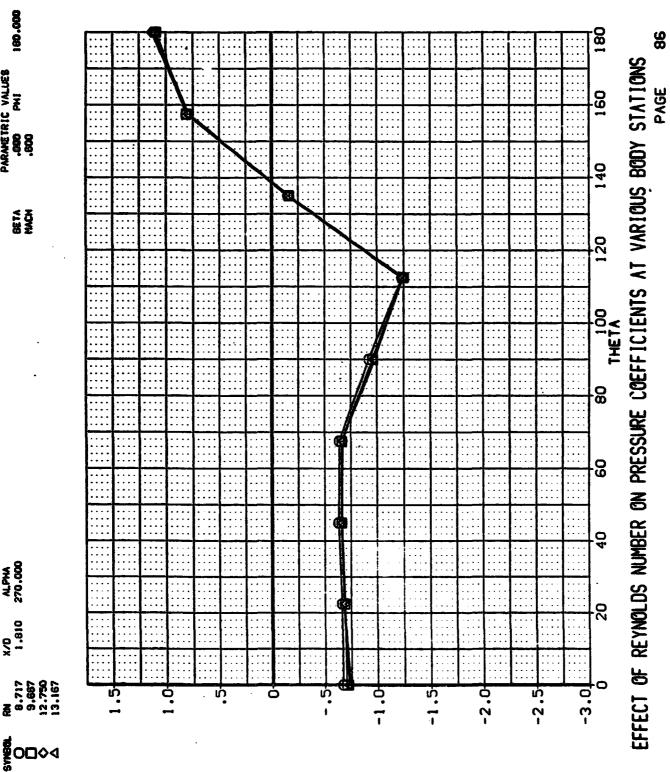


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PARMETRIC VALUES .000 PHI .800 (B1E106) MSFC HRWT 033.(SA29F) SRB FORWARD NOSE SECTION ALPHA 270.000 1.810

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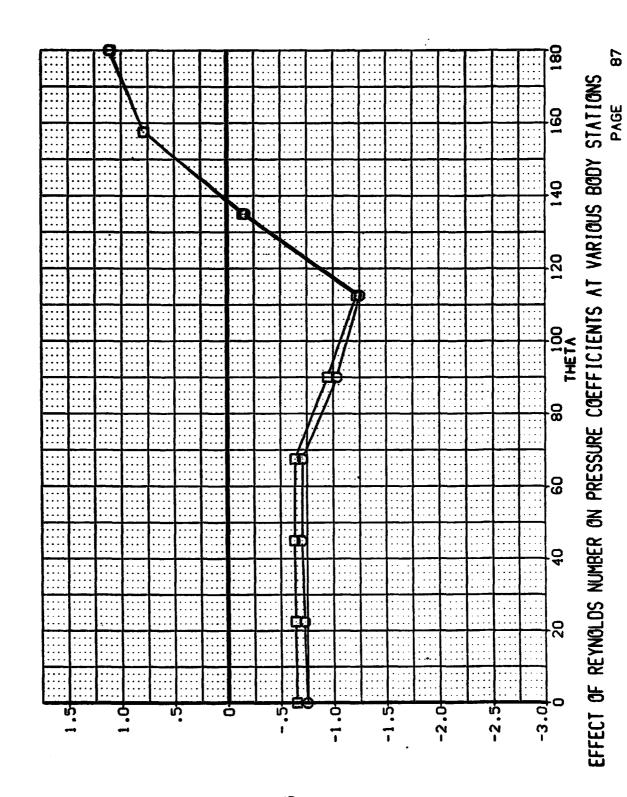


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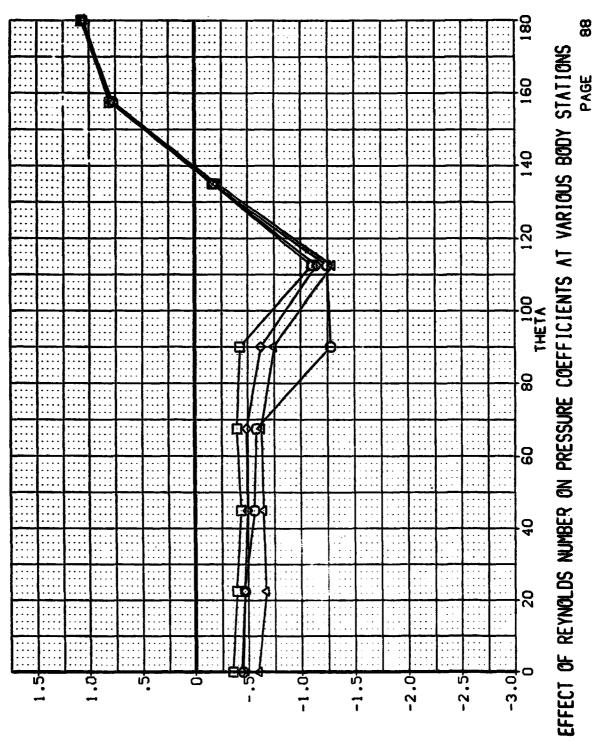
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EFFECT OF REYNOLDS NUMBER ON PRESSURE COEFFICIENTS AT VARIOUS BODY STATIONS

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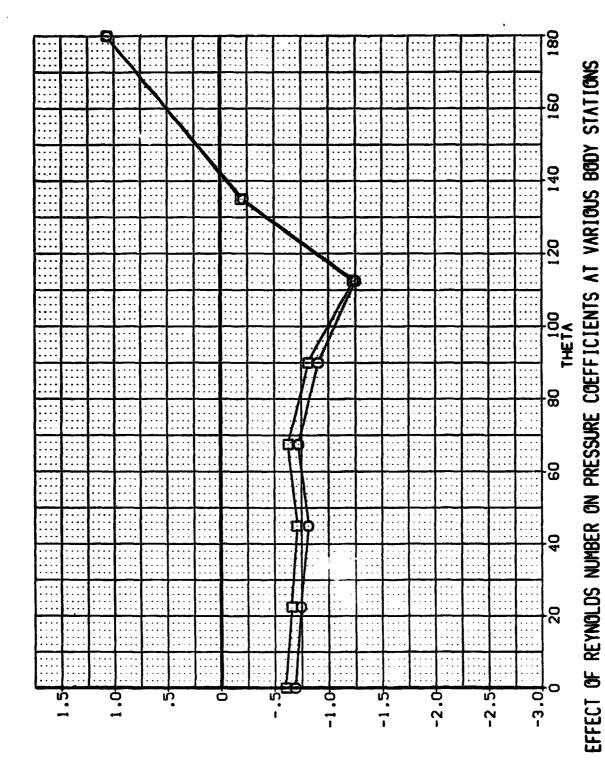
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APPENDIX TABULATED SOURCE DATA

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19 FEB 76	

	(MEIDD) (ID DEC /D /	PARAMETRIC DATA		8 - 2.2500 PO - 67.800				1 - 4.0167 PO - 128.12		•	
DATE 19 FEB 76 TABULATED SOURCE DATA, MSFC HRNT 033 (SARBF)	MSFC MRWI USS (SARWF) SHB FORWARD NOSE SECTION	REFERENCE DATA	SREF = 116.2800 SQ.FT. XMRP = 1044.0000 IN. XS LREF = 146.0000 IN. YMRP = .0000 IN. YS MACH SPEF = 146.0000 IN. ZMRP = .0000 IN. ZS SCALE = .0137	P: (1) = 2.250 ALPHA (1) = 90.000 Q(PSI) = 7.2900 TO = 37.000 RN	SECTION (1) SRB NOSE DEPENDENT VARIABLE CP	x,D 1.9100 2.1600	71#E1A .03029703280 22.50034605040 45.00029506170 57.50036003810 12.5009110 -1.110 135.00019303280 157.530 .7770 .7130	PN (2) = 4.017 ALPHA (1) = 270.000 Q(PSI) = 13.370 TO = 51.000 RN	SECTICAL (1) SAB NOSE CEPENDENT VARIABLE CP	X/D 1.8100 2.1600	1HETA 8240 3290 3290 92.500

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DATE 19 FEB 76					TAE	BULATED	SOURCE (MATA	ABULATED SOURCE DATA, MSFC HRWT 033 (SA29F)	WT 033 (SA2B	£					_	PAGE	13
					_	MSFC HRW	T 033 (9	\$ 4 29	MSFC HRWI 033 (SA29F) SRB FORWARD NOSE SECTION	FRWARD NO	35 S	ECTION	7			(RIE105)			
RN (3) •	6.067	ALPHA	_	_	n	270.000 G(PSI) = 19.930	0(PS1)	0	19.930	5	n	46.000	8	Z	•	6.0867	2		191.41
SECTION 1 11SRB	NCSE					DEPE	DEPENDENT VARIABLE CP	RIA	BLE CP										
X/D 1.8100	00 2.1600	_																	
7HETA - 000 4970 22.500 4230 45.000 3560 67.500 - 1. 0390 112.500 - 1. 1230 135.000 - 1. 1240 157.500 - 1. 1240 167.500 - 1. 1240													;						
RN (+)	7.517	AL PHA	-	9		270.000 G(PS1) = 25.360	0(PS1)	0	23.360	5	ø	49.000	8	Æ	H .	7.5167	8		238.79
SECTION (1) SRB	NOSE					DEPE	DEPENDENT VARIABLE CP	RIA	BLE CP										
x/0 1.8100	0 2.1600																		
7HETA .0006830 22.5003710 45.0003770 67.5003590 90.000 -1.1760 112.500 -1.0976 135.000 -1.0870 157.500 -1.0870	2810 27440 27440 27440 27440 27440 27440 27440 27440 27440 2740 27																		
RN (5) #	9.017	ALPHA	_	-		270.000 Q(PSI) =	Q(PSL)	0	30.220	0		49.000	8	Æ	8	9.0187	8	8	286.83
SECTION (1) SRB	NOSE					DEPE	DEPENDENT VARIABLE CP	RIA	BLE CP										
x/D 1.8100	0 2.1600	_																	
7HETA .000 - 4970 22.500 - 4270 45.000 - 4190 67.500 - 1190 112.500 - 1.1690 135.000 - 2150 157.500 - 2150 160.000 1.000	103990 104900 104490 10 - 1 .650 10 - 1 .1530 10 - 1 .3070 10 - 3070 103070															·	•		

DATE 19 FEB 76	TABU	ATEO S	URCE DAT	TABULATED SOURCE DATA, MSFC HRWT 033 (SA29F)	IT 033 (9	:A29F)					PAGE	±
	ξ.	C HRWT	033 (SAE	MSFC HRWT 033 (SAP9F) SRB FORWARD NOSE SECTION	MARD NO	3E 3EC	NOIT		(RIE105)			
RN (6) = 10.217 ALPHA (1)	u	0.000	0(PSI) =	90.000 Q(PSI) = 32.730	5	0	39.000	ž	10.217	8	8	315.88
SECTION (1) SRB NOSE		DEPEN	DEPENDENT VARIABLE CP	ABLE CP								
X/D 1.8100 2.1600												
THETA .00056204110 22.50041205210 45.00044105560 67.50041604710 90.0001270 -1.0880 112.50083203320 157.500 .7590 .7240												
RN (7) = 13.383 ALPHA (1)	0	0.000	@(PSI) =	270.000 Q(PSI) = 44.620	5	<i>3</i>	48.000	ž	13.383	8	¥	424.20
SECTION (1) SRB NOSE		DEPEN	DEPENDENT VARIABLE CP	ABLE CP								
X/D 1.8100 2.1600												
THETA498038805880450049804910												·
RN (8) = 18.800 ALPHA (1)		000.0	- (184)0	270.000 O(PSI) = 61.330	5	3	44.000	ž	19.800	8	5	590.08
SECTION (1) SAB NOSE		DEPER	DEPENDENT VARIABLE CP	ABLE CP								
X/D 1.8100 2.1500												
74E1A .00057804420 22.50051505190 45.00044104950 67.50049105000 97.000 -1.3490 -1.3560 135.00027403500 157.592 .1340 .6950												

PAGE 15	(RIE106) (15 DEC 75)	PARAMETRIC DATA	A • .000 PH1 • 180.000	RN = 2.7667 PO = 54.300				FN = 4.0500 PO = 92.220		•	
TABUL / TED SOURCE DATA, MSFC HRWT 033 (SAZGF)	MSFC HRWT 033 (SA29F) SRB FORMARD NOSE SECTION	DATA	7. XMRP = 1044.0000 IN. XS BETA MACH YMRP = .0000 IN. YS MACH ZMRP = .0000 IN. ZS	ALPHA (1) = 270.000 Q(PSI) = 12.990 TO = 48.000	DEPENDENT VARIABLE CP	01		ALPHA (1) = 270.000 Q(PSI) = 18.680 TO = 41.000	DEPENDENT VARIABLE CP	01	
DATE 19 FEB 76		REFERENCE DAT	SREF = 116.2600 SQ.FT. LREF = 146.0000 IN. BREF = 145.0000 IN. SCALE = .0137	RN (1) = 2.767	SECTION (1) SAB NOSE	X,D 1.8100 2.1600	11-ETA .00051504450 22.50045904710 45.00045605830 67.50044505820 90.00044505820 112.5001460 -1.2770 135.0001140 -1.750 157.50011401760 157.50011401760 157.50011401760	RN (2) = 4.050	SECTION (11SPB NOSE	x/0 1.8:00 2.1600	THEIA

FEB 76 TABULATED SOURC	Tabulated Source Data, MSFC Hant 033 (Sa29F)	T 033 (S.	(362)					T.	PAGE	9
MSFC HRWT 033	MSFC HRWT 033 (SA29F) SRB FORWARD NOSE SECTION	MARD NOS	: SECT1(z			(RIE106)			
5.850 ALPHA (1) = 270.000 Q(F	0(PSI) = 27.260	5	a 46.000	9	Z	0	5.6500	8	e e	136.57
NOSE DEPENDENT	DEPENDENT VARIABLE CP									
00 2.1600										
- 4350 - 4700 - 4930 - 6240 - 1540 - 1770 - 7890 1.0600										
7.417 ALPHA (1) = 270,000 G(P	0(PSI) = 34.410	10	51.000	000	Æ		7.4167	8		175.40
SAB NOSE DEPENDENT	DEPENDENT VARIABLE CP									
.8100 2.1600										
65205930 64706690 61906430 62606350 23207470 15802140 15802140 0550 1.0590				·						
8.717 ALPHA (1) = 270.000 G(P	0(PSI) = 40.290	10	= 47.000	000	æ		9.7167	8	0	204.15
NOSE DEPENDENT	DEPENDENT VARIABLE CP				•					
100 2.1600										
.66905390 .65805700 .63406590 .63406590 .5502500 .5502090 .5502090										

19 FEB 76 TABULATE MSFC H	'ABULATED SOURCE DATA, MSFC HRWT 033 (SA29F) MSFC HRWT 033 (SA29F) SAB FORWARD NOSE SECTION	IT 033 (SA29F)	<u>8</u>	(R1E106)	106)	PAGE 17
9.667 ALPMA (1) = 270.00	270.000 Q(PSI) = 44.820	10 = 51	51.000 FRN	€ 1	8	. 229.85
1) SAB NOSE DE	DEPENDENT VARIABLE CP					
1.8100 2.1600						
70406390 68907170 68207040 66407000 15901890 15902110 15902110						
= 12.750 ALPHA (1) = 270.00	270.000 0(PSI) = 59.920	TO = 47	47.000 RM	- 12.750	8	. 298.82
1) SRB NOSE DE	DEPENDENT VARIABLE CP					
1.8100 2.1600						
72406230 68707040 66307710 96506720 35508210 18202030 7910 .7320	•					
= 13.167 ALPHA (1) = 270.00	270.000 Q(PSI) = 60.110	170 = 41	41.000 FRN	= 13.167	8	* 303.58
11SRB NOSE DE	DEPENDENT VARIABLE CP					٠
1.8100 2.1600						
74206370 68006820 65907650 98008130 -1.2280 -1.2230 14801950 7910 -2.2760						

ORIGINAL PAGE IS OF POOR QUALITY

DATE 19 FEB	EB 76			_ _	TABULATED SOURCE DATA, MSFC HRWT 033 (SA28F)	OURCE DAT	A. MSFC 1#	WT 033	SAZEF	_				•	PAGE	
					MSFC HRWT 033 (SA29F) SAB FOFTARD NOSE SECTION	033 (SA2	9F) SAB F(SP-LARO N	35 350	NO I			(RIE106)			
S N		16.083	ALPHA (1)		270.000 Q(PSI) = 74.630	0(PSI) •	74.630	5		49.000	Ę	=	18.083	8	19	378.90
SECTION (1) SRB NOSE			DEPEN	DEPENDENT VARIABLE CP	ABLE CP									
Q/x	. .	.8100 2.1600														
THE TA . 000	7															
45.000 67.500 90.000	7030															
135.500 135.900 157.500	-1.25¥0 -1.1710 -78¥0	540 -1.2550 7102120 840 -3.1300 140 1.0600														
NR 010		20.050	ALFHA		270.000 Q(PSI) = 90.350	0(PS1)	90.350	5	•	• 36.000	£	ā.	20.050	8	•	• 451.37
SECTION (11.56	115RB NOSE			DEPEN	DEPENDENT VARIABLE CP	ABLE CP									
O/x	1.8	.810n 2.16CJ														
##TA - 000 - 22.500 - 62.500 - 62.500 - 62.500 - 62.500 - 63.500 - 63.5000 - 63.500 - 63.500 - 63.500 - 63.500 - 63.500 - 63.500 - 63.5	- 64480 - 6440 - 6440 - 6440 - 6440 - 6440 - 6460 -	+806040 5307100 5307100 5308310 5308310 5301910 5101910 570 - 3.7420 570 - 3.7420														